# Approach to Calculate the Global Climate Change: $+0.8^{\circ} \mathrm{C}$ for the Past Century 

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#### Abstract

In analogy with the ultimate speed $c$, there is an ultimate acceleration $\beta$, nobody's acceleration can exceed this limit $\beta$, in the solar system, $\beta=2.961520 \mathrm{e}+10\left(\mathrm{~m} / \mathrm{s}^{2}\right)$. Because this ultimate acceleration is a large number, any effect connecting to $\beta$ will become easy to test, including quantum gravity tests. In this paper, an approach is put forward to connect the ultimate acceleration with quantum theory, as an application, this relativistic model gives out the sunspot cycle to be 10.38 years due to the ultimate acceleration. The same approach is applied to the Earth El Niño oscillation and global climate change cycle, calculating out the El Niño irregular period to be 8.5 years; calculating out the global climate change cycle to be 100.6 kiloyears which coincides with the Milankovitch cycle; the global temperature has increased about $0.8^{\circ} \mathrm{C}$ for the past century.


## 1. Introduction

In general, some quantum gravity proposals $[1,2]$ are extremely hard to test in practice, as quantum gravitational effects are appreciable only at the Planck scale [3]. But ultimate acceleration gives another scheme to enhance the quantum gravity effects.

In 1843, German pharmacist H.S. Schwabe found that the growth and decline of sunspots had a period of about 10 years through his own observation records of sunspots for more than 20 years. In 1848 , J.R. Wolf calculated the monthly sunspot relative number to 1749 , thus affirming that the average cycle length of sunspot activity was 11.1 years. In 2019, there are 288 days without sunspots, and 40 consecutive days ahead. On December 25,2019 , two new sunspots appeared in the southern hemisphere of the sun, indicating the arrival of the 25 th sunspot cycle, which is expected to reach its peak in July 2025. Although it is known that the cycle of sunspot activity is about 11 years, what drives this non-stop pattern is still a mystery.

Detailed estimates of temperature and carbon dioxide levels over the past 800000 years can be made from Antarctic ice samples. Within this time range, Earth's temperature has gone up and down, with glaciation and $\mathrm{CO}_{2}$ levels [32]. Although broadly consistent with the Milankovitch hypothesis, a number of specific observations are not explained by the hypothesis.

In analogy with the ultimate speed $c$, there is an ultimate acceleration $\beta$, nobody's acceleration can exceed this limit $\beta$, in the solar system, $\beta=2.961520 \mathrm{e}+10\left(\mathrm{~m} / \mathrm{s}^{2}\right)$. Because this ultimate acceleration is a large number, any effect connecting to $\beta$ will become easy to test, including quantum gravity tests. In this paper, an approach is put forward to connect the ultimate acceleration with quantum theory, as an application, this relativistic model gives out the sunspot cycle to be
10.38 years due to the ultimate acceleration. The same approach is applied to the Earth El Niño oscillation and global climate change cycle, calculating out the El Niño irregular period to be 8.5 years; calculating out the global climate change cycle to be 100.6 kiloyears which coincides with the Milankovitch cycle; the global temperature has increased about $0.8^{\circ} \mathrm{C}$ for the past century.

## 2. How to connect the ultimate acceleration with quantum theory

In the relativity, the speed of light $c$ is an ultimate speed, nobody's speed can exceed this limit $c$. The relativistic velocity $u$ of a particle in the coordinate system $\left(x_{1}, x_{2}, x_{3}, x_{4}=i c t\right)$ satisfies

$$
\begin{equation*}
u_{1}^{2}+u_{2}^{2}+u_{3}^{2}+u_{4}^{2}=-c^{2} \tag{1}
\end{equation*}
$$

No matter what particles (electrons, molecules, neutrons, quarks), their 4-vector velocities all have the same magnitude: $|u|=i c$. All particles gain equality because of the same magnitude of the 4velocity $u$. The acceleration $a$ of a particle is given by

$$
\begin{equation*}
a_{1}^{2}+a_{2}^{2}+a_{3}^{2}=a^{2} ; \quad\left(a_{4}=0 ; \quad \because x_{4}=i c t\right) \tag{2}
\end{equation*}
$$

Assume that particles have an ultimate acceleration $\beta$ as limit, no particle can exceed this acceleration limit $\beta$. Subtracting the both sides of the above equation by $\beta^{2}$, we have

$$
\begin{equation*}
a_{1}^{2}+a_{2}^{2}+a_{3}^{2}-\beta^{2}=a^{2}-\beta^{2} ; \quad a_{4}=0 \tag{3}
\end{equation*}
$$

It can be rewritten as

$$
\begin{equation*}
\left[a_{1}^{2}+a_{2}^{2}+a_{3}^{2}+0+(i \beta)^{2}\right] \frac{1}{1-a^{2} / \beta^{2}}=-\beta^{2} \tag{4}
\end{equation*}
$$

Now, the particle subjects to an acceleration whose five components are specified by

$$
\begin{array}{ll}
\alpha_{1}=\frac{a_{1}}{\sqrt{1-a^{2} / \beta^{2}}} ; \quad \alpha_{2}=\frac{a_{2}}{\sqrt{1-a^{2} / \beta^{2}}}  \tag{5}\\
\alpha_{3}=\frac{a_{3}}{\sqrt{1-a^{2} / \beta^{2}}} ; \quad \alpha_{4}=0 ; \quad \alpha_{5}=\frac{i \beta}{\sqrt{1-a^{2} / \beta^{2}}}
\end{array}
$$

where $\alpha_{5}$ is the newly defined acceleration in five dimensional space-time ( $x_{1}, x_{2}, x_{3}, x_{4}=i c t, x_{5}$ ). Thus, we have

$$
\begin{equation*}
\alpha_{1}^{2}+\alpha_{2}^{2}+\alpha_{3}^{2}+\alpha_{4}^{2}+\alpha_{5}^{2}=-\beta^{2} ; \quad \alpha_{4}=0 \tag{6}
\end{equation*}
$$

It means that the magnitude of the newly defined acceleration $\alpha$ for every particle takes the same value: $|\alpha|=i \beta$ (constant imaginary number), all particle accelerations gain equality for the sake of the same magnitude.

How to resolve the velocity $u$ and acceleration $\alpha$ into $x, y$, and $z$ components? In realistic world, a hand can rotate a ball moving around a circular path at constant speed $v$ with constant centripetal acceleration $a$, as shown in Fig.1(a).


Fig. $1 \quad$ (a) A hand rotates a ball moving around a circular path at constant speed $v$ with constant centripetal acceleration $a$. (b) The particle moves along the $x_{I}$ axis with the constant speed $|u|=i c$ in the $u$ direction and constant centripetal force in the $x_{5}$ axis at the radius $i R$ (imaginary number).
$<$ Clet2020 Script>//[26]
double $\mathrm{D}[100], \mathrm{S}[2000]$;int $\mathrm{i}, \mathrm{j}, \mathrm{R}, \mathrm{X}, \mathrm{N}$;
int main() $\{\mathrm{R}=50 ; \mathrm{X}=50 ; \mathrm{N}=600 ; \mathrm{D}[0]=-50 ; \mathrm{D}[1]=0 ; \mathrm{D}[2]=0 ; \mathrm{D}[3]=\mathrm{X} ; \mathrm{D}[4]=0 ; \mathrm{D}[5]=0 ; \mathrm{D}[6]=-50 ; \mathrm{D}[7]=\mathrm{R} ; \mathrm{D}[8]=0$;
$\mathrm{D}[9]=600 ; \mathrm{D}[10]=10 ; \mathrm{D}[11]=\mathrm{R} ; \mathrm{D}[12]=0 ; \mathrm{D}[13]=3645$;
Lattice(SPIRAL,D,S);SetViewAngle( $0,80,-50$ );DrawFrame(FRAME NULL, 1,0xffffff);
Draw("LINE, 0,2,XYZ, 0 ","-150,0,0,-50,0,0");Draw("ARROW,0,2,XȲZ, $10 ", " 50,0,0,150,0,0 ")$;
SetPen(2,0xff0000);Plot("POLYLINE,0,600,XYZ",S[9]);i=9+3*N-6;Draw("ARROW,0,2,XYZ,10",S[i]);
TextHang(S[i],S[i+1],S[i+2]," \#if|u|=ic\#t");TextHang(150,0,0," \#ifx\#sd1\#t");SetPen(2,0x005fff);
Draw("LINE, 1,2,XYZ,8","-50,0,50,-50,0,100");Draw("LINE, 1,2,XYZ,8","-40,0,50,-40,0,100");
Draw("ARROW,0,2,XYZ,10","-80,0,100,-50,0,100");Draw("ARROW,0,2,XYZ, 10","-10,0,100,-40,0,100");
TextHang (-50, $0,110, " 1$ spiral step"); $1=9+3 * N ; S[i]=50 ; S[i+1]=10 ; S[i+2]=10$;
Draw("ARROW,0,2,XYZ, 10 ","50,0,0,50,80,80");TextHang(50,80,80," \#ifx\#sd5\#t");
Draw("ARROW,0,2,XYZ, 10 "," $50,72,0,50,0,72$ "); TextHang(50,0,72," \#ifx\#sd4\#t");
SetPen(3,0x00ffff);Draw("ARROW,0,2,XYZ,15",S[i-3]);TextHang(S[i],S[i+1],S[i+2]," \#if| $\alpha \mid=i \beta \# t ")$;
SetPen(3,0x00ff00);Draw("ARROW,0,2,XYZ,15","50,0,0,120,0,0");TextHang(110,5,5," \#ifJ\#t");
TextHang(-60,0,-80," right hand chirality"); $\} \# v 07=?>A$

In analogy with the ball in a circular path, consider a particle in one dimensional motion along the $x_{1}$ axis at the speed $v$, in the Fig.1(b) it moves with the constant speed $|u|=i c$ almost along the $x_{4}$ axis and slightly along the $x_{1}$ axis, and the constant centripetal acceleration $|\alpha|=i \beta$ in the $x_{5}$ axis at the constant radius $i R$ (imaginary number); the coordinate system ( $x_{l}, x_{4}=i c t, x_{5}=i R$ ) establishes a cylinder coordinate system in which this particle moves spirally at the speed $v$ along the $x_{1}$ axis. According to usual centripetal acceleration formula $a=v^{2} / r$, the acceleration in the $x_{4}-x_{5}$ plane is given by

$$
\begin{equation*}
a=\frac{v^{2}}{r} \Rightarrow i \beta=\frac{|u|^{2}}{i R}=-\frac{c^{2}}{i R}=i \frac{c^{2}}{R} \tag{7}
\end{equation*}
$$

Therefore, the track of the particle in the cylinder coordinate system $\left(x_{1}, x_{4}=i c t, x_{5}=i R\right)$ forms a shape, called as acceleration-roll. The faster the particle moves along the $x_{I}$ axis, the longer the spiral step is.

As like a steel spring that contains elastic wave, the track in the acceleration-roll in Fig.1(b) can be described by a wave function whose phase changes $2 \pi$ for one spiral step. Apparently, this wave is the de Broglie's matter wave for electrons, protons or quarks, etc.

Proof: The wave function phase changes $2 \pi$ for one spiral circumference $2 \pi(i R)$, then a small displacement of the particle on the spiral track is $|u| d \tau=i c d \tau$ in the 4 -vector $u$ direction, thus this wave phase along the spiral track is evaluated by

$$
\begin{equation*}
\text { phase }=\int_{0}^{\tau} \frac{2 \pi}{2 \pi(i R)} i c d \tau=\int_{0}^{\tau} \frac{c}{R} d \tau \tag{8}
\end{equation*}
$$

Substituting the radius $R$ into it, the wave function $\psi$ is given by

$$
\begin{equation*}
\psi=\exp (-i \cdot \text { phase })=\exp \left(-i \int_{0}^{\tau} \frac{c}{R} d \tau\right)=\exp \left(-i \frac{\beta}{c} \int_{0}^{\tau} d \tau\right) \tag{9}
\end{equation*}
$$

In the theory of relativity, we known that the integral along $d \tau$ needs to transform into realistic line integral, that is

$$
\begin{align*}
& d \tau=-c^{2} \frac{d \tau}{-c^{2}}=\left(u_{1}^{2}+u_{2}^{2}+u_{3}^{2}+u_{4}^{2}\right) \frac{d \tau}{-c^{2}} \\
& =\left(u_{1} d x_{1}+u_{2} d x_{2}+u_{3} d x_{3}+u_{4} d x_{4}\right) \frac{1}{-c^{2}} \tag{10}
\end{align*}
$$

Therefore, the wave function $\psi$ is evaluated by

$$
\begin{align*}
& \psi=\exp \left(-i \frac{\beta}{c} \int_{0}^{\tau} d \tau\right)  \tag{11}\\
& =\exp \left(i \frac{\beta}{c^{3}} \int_{0}^{x}\left(u_{1} d x_{1}+u_{2} d x_{2}+u_{3} d x_{3}+u_{4} d x_{4}\right)\right)
\end{align*}
$$

This wave function may have different explanations, depending on the particle under investigation. If the $\beta$ is replaced by the Planck constant, the wave function of electrons is given by

$$
\begin{align*}
& \text { assume: } \quad \beta=\frac{m c^{3}}{\hbar}  \tag{12}\\
& \psi=\exp \left(\frac{i}{\hbar} \int_{0}^{x}\left(m u_{1} d x_{1}+m u_{2} d x_{2}+m u_{3} d x_{3}+m u_{4} d x_{4}\right)\right)
\end{align*}
$$

where $m u_{4} d x_{4}=-E d t$, it strongly suggests that the wave function is just the de Broglie's matter wave [4,5,6].

In Fig.1(b), the acceleration-roll of particle moves with two distinctions: right-hand chirality and left-hand chirality. The direction of the angular momentum $J$ would be slightly different from the $x_{1}$ due to spiral precession. It is easy to calculate the ultimate acceleration $\beta$, the radius $R$ and the angular momentum $J$ in the plane $x_{4}-x_{5}$ for a spiraling electron as

$$
\begin{align*}
& \beta=\frac{c^{3} m}{\hbar}=2.327421 \mathrm{e}+29\left(\mathrm{M} / \mathrm{s}^{2}\right) \\
& R=\frac{c^{2}}{\beta}=3.861593 \mathrm{e}-13(\mathrm{M})  \tag{13}\\
& J= \pm m|u| i R=\mp \hbar
\end{align*}
$$

$<$ Clet2020 Script>//[26]
double beta,R,J,m,D[10];char str[200];
int main() $\{\mathrm{m}=\mathrm{ME} ;$ beta=SPEEDC*SPEEDC*SPEEDC*m/PLANCKBAR;
$\mathrm{R}=$ SPEEDC*SPEEDC/beta;J=PLANCKBAR;Format(str,"beta $=\% \mathrm{e}, \# \mathrm{nR}=\% \mathrm{e}, \# \mathrm{~nJ}=\% \mathrm{e}$ ", beta, $\mathrm{R}, \mathrm{J}$ );
TextAt(50,50,str);ClipJob(APPEND,str); $\} \# v 07=\# t$

Considering another explanation to $\psi$ for planets in the solar system, no Planck constant can be involved. But, in a many-body system with the total mass $M$, the data-analysis [28] tells us that the ultimate acceleration can be rewritten in terms of Planck-constant-like constant $h$ as

$$
\begin{align*}
& \text { assume: } \quad \beta=\frac{c^{3}}{h M}  \tag{14}\\
& \psi=\exp \left(\frac{i}{h M} \int_{0}^{x}\left(u_{1} d x_{1}+u_{2} d x_{2}+u_{3} d x_{3}+u_{4} d x_{4}\right)\right)
\end{align*}
$$

The constant $h$ will be determined by experimental observations. This paper will show that this wave function is applicable to several many-body systems in the solar system, the wave function is called as the acceleration-roll wave.

Considering third explanation to $\psi$ for atoms in cells and viruses. Typically, sound speed in water is $v=1450 \mathrm{~m} / \mathrm{s}$, according to $\nu=f \lambda$, the sound wavelength $\lambda$ for frequency $1 \mathrm{kHz}---1 \mathrm{Mhz}$ is $1.5 \mathrm{~m}---1.45 \mathrm{~mm}$. In general, the sound wavelength is larger than cell size, because cell size is about 1 micro. Thus, almost all cells and viruses live in a smaller space which is not sensitive to the sound. The acceleration-roll can provide a kind of wave with a shorter wavelength and lower frequency for various cells and viruses beyond human-sensitive sound wave [28].
Of cause, there is an "ultimate distance" [28] from which the Hubble law can be derived out.
Tip: actually, ones cannot get to see the acceleration-roll of particle in the relativistic spacetime $\left(x_{1}, x_{2}, x_{3}, x_{4}=i c t\right)$; only get to see it in the cylinder coordinate system $\left(x_{1}, x_{4}=i c t, x_{5}=i R\right)$.

## 3. How to determine the ultimate acceleration

In the Bohr's orbit model for planets or satellites, as shown in Fig.2, the circular quantization condition is given in terms of relativistic matter wave in gravity by

$$
\left.\begin{array}{c}
\frac{\beta}{c^{3}} \oint_{L} v_{l} d l=2 \pi n  \tag{15}\\
v_{l}=\sqrt{\frac{G M}{r}}
\end{array}\right\} \Rightarrow \sqrt{r}=\frac{c^{3}}{\beta \sqrt{G M}} n ; n=0,1,2, \ldots
$$



Fig. 2 A planet 2D orbit around the sun, an acceleration-roll winding around the planet.
<Clet2020 Script>//[26]
int i,j,k; double r,rot,x,y,z,D[20],F[20],S[200]; int main()\{SetViewAngle("temp0, theta60,phi-30");
DrawFrame(FRAME_LINE, 1,0xafffaf);r=80;Spiral(); TextHang(r,-r,0,"acceleration-roll");
r=110;TextHang(r,0,0,"x");TextHang(0,r,0,"y");TextHang(0,0,r,"z");\}
Spiral () $\{\mathrm{r}=80 ; \mathrm{j}=10 ;$ rot $=\mathrm{j} / \mathrm{r} ; \mathrm{k}=2 * \mathrm{PI} /$ rot +1 ;
for $(\mathrm{i}=0 ; \mathrm{i}<\mathrm{k} ; \mathrm{i}+=1)\{\mathrm{D}[0]=\mathrm{x} ; \mathrm{D}[1]=\mathrm{y} ; \mathrm{D}[2]=\mathrm{z} ; \mathrm{D}[6]=\mathrm{x} ; \mathrm{D}[7]=\mathrm{y} ; \mathrm{D}[8]=\mathrm{r}$;
$\mathrm{x}=\mathrm{r}^{*} \cos ($ rot $* \mathrm{i}) ; \mathrm{y}=\mathrm{r} * \sin \left(\right.$ rot $\left.^{*} \mathrm{i}\right) ; \mathrm{z}=0 ; \mathrm{if}(\mathrm{i}==0)$ continue;
$\operatorname{SetPen}(2,0 x 00) ; F[0]=D[0] ; F[1]=D[1] ; F[2]=x ; F[3]=y ; D r a w(" L I N E, 0,2, X Y, ", F) ; S e t P e n(1,0 x f f 0000) ;$
$\mathrm{D}[3]=\mathrm{x} ; \mathrm{D}[4]=\mathrm{y} ; \mathrm{D}[5]=\mathrm{z} ; \mathrm{D}[9]=40 ; \mathrm{D}[10]=10 ; \mathrm{D}[11]=8 ; \mathrm{D}[12]=0 ; \mathrm{D}[13]=360$;
Lattice(SPIRAL,D,S);Plot("POLYLINE, $0,40, X Y Z ", S[9]) ;\}$
\} $\# \mathrm{v} 07=$ ? $>\mathrm{A}$

The solar system, Jupiter's satellites, Saturn's satellites, Uranus' satellites, Neptune's satellites as five different many-body systems are investigated with the Bohr's orbit model. After fitting observational data as shown in Fig.3, their ultimate accelerations are obtained in Table 1. The predicted quantization-blue-lines in Fig.3(a), Fig.3(b), Fig.3(c), Fig.3(d) and Fig.3(e) agree well with experimental observations for those inner constituent planets or satellites.


Fig. 3 The orbital radii are quantized for inner constituents. (a) the solar system with $h=4.574635 \mathrm{e}-16\left(\mathrm{~m}^{2} \mathrm{~s}^{-1} \mathrm{~kg}^{-1}\right)$.

The relative error is less than $3.9 \%$. (b) the Jupiter system with $h=3.531903 \mathrm{e}-16\left(\mathrm{~m}^{2} \mathrm{~s}^{-1} \mathrm{~kg}^{-1}\right)$. Metis and Adrastea are assigned the same quantum number for their almost same radius. The relative error is less than $1.9 \%$. (c) the Saturn system with $h=6.610920 \mathrm{e}-16\left(\mathrm{~m}^{2} \mathrm{~s}^{-1} \mathrm{~kg}^{-1}\right)$. The relative error is less than $1.1 \%$. (d) the Uranus system with $h=1.567124 \mathrm{e}-16\left(\mathrm{~m}^{2} \mathrm{~s}^{-1} \mathrm{~kg}^{-1}\right) . n=0$ is assigned to the Uranus. The relative error is less than $2.5 \%$. (e) the Neptune system with $h=1.277170 \mathrm{e}-16\left(\mathrm{~m}^{2} \mathrm{~s}^{-1} \mathrm{~kg}^{-1}\right) . n=0$ is assigned to the Neptune. The relative error is less than $0.17 \%$.

Table 1 Planck-constant-like constant $h, \mathrm{~N}$ is constituent particle number with smaller orbital inclination.

| system | N | $M / M_{\text {earth }}$ | $\beta(\mathrm{m} / \mathrm{s} 2)$ | $h(\mathrm{~m} 2 \mathrm{~s}-1 \mathrm{~kg}-1)$ | Prediction |
| :---: | :---: | :---: | :---: | :---: | :---: |
| Solar planets | 9 | 333000 | $2.961520 \mathrm{e}+10$ | $4.574635 \mathrm{e}-16$ | Fig.3(a) |
| Jupiter' satellites | 7 | 318 | $4.016793 \mathrm{e}+13$ | $3.531903 \mathrm{e}-16$ | Fig.3(b) |
| Saturn's satellites | 7 | 95 | $7.183397 \mathrm{e}+13$ | $6.610920 \mathrm{e}-16$ | Fig.3(c) |
| Uranus' satellites | 18 | 14.5 | $1.985382 \mathrm{e}+15$ | $1.567124 \mathrm{e}-16$ | Fig.3(d) |
| Neptune 's satellites | 7 | 17 | $2.077868 \mathrm{e}+15$ | $1.277170 \mathrm{e}-16$ | Fig.3(e) |

Besides every $\beta$, our interest shifts to the constant $h$ in Table 1 , which is defined as

$$
\begin{equation*}
h=\frac{c^{3}}{M \beta} \Rightarrow \sqrt{r}=h \sqrt{\frac{M}{G}} n \tag{16}
\end{equation*}
$$

In a many-body system with the total mass $M$, a constituent particle has the mass $m$ and moves at the speed $v$, it is easy to find that the wavelength of de Broglie's matter wave should be modified for planets and satellites as

$$
\begin{equation*}
\lambda_{d e_{-} B r o g l i e}=\frac{2 \pi \hbar}{m v} \Rightarrow \text { modify } \Rightarrow \lambda=\frac{2 \pi h M}{v} \tag{17}
\end{equation*}
$$

where $h$ is a Planck-constant-like constant. Usually the total mass $M$ is approximately equal to the central-star's mass. It is found that this modified matter wave works for quantizing orbits correctly in Fig. 3 [28,29]. The key point is that the various systems have almost same Planck-constant-like constant $h$ in Table 1 with a mean value of $3.51 \mathrm{e}-16 \mathrm{~m}^{2} \mathrm{~s}^{-1} \mathrm{~kg}^{-1}$, at least have the same magnitude! The acceleration-roll wave is a generalized matter wave as a planetary scale wave.

In Fig.3(a), the blue straight line expresses the linear regression relation among the Sun, Mercury, Venus, Earth and Mars, their quantization parameters are $h M=9.098031 \mathrm{e}+14\left(\mathrm{~m}^{2} / \mathrm{s}\right)$. The ultimate acceleration is fitted out to be $\beta=2.961520 \mathrm{e}+10\left(\mathrm{~m} / \mathrm{s}^{2}\right)$. Where, $n=3,4,5, .$. were assigned to solar planets, the sun was assigned a quantum number $n=0$ because the sun is in the central state.

## 4. Optical model of the central state

The acceleration-roll wave as the relativistic matter wave generalized in gravity is given by

$$
\begin{equation*}
\psi=\exp \left(\frac{i}{h M} \int_{0}^{x} v_{l} d l\right) ; \quad \lambda=\frac{2 \pi h M}{v_{l}}=\frac{2 \pi c^{3}}{v_{l} \beta} . \tag{18}
\end{equation*}
$$

In a central state $n=0$, if the coherent length of the acceleration-roll wave is long enough, its head may overlap with its tail when the particle moves in a closed orbit in the space time, as shown in Fig.4, the interference of the acceleration-roll wave between its head and tail will occur in the overlapping zone. The overlapped wave is given by

$$
\begin{align*}
& \psi(r)=1+e^{i \delta}+e^{i 2 \delta}+\ldots+e^{i(N-1) \delta}=\frac{1-\exp (i N \delta)}{1-\exp (i \delta)} \\
& \delta(r)=\frac{1}{h M} \oint_{L}\left(v_{l}\right) d l=\frac{2 \pi \omega r^{2}}{h M}=\frac{2 \pi \beta \omega r^{2}}{c^{3}} \tag{19}
\end{align*}
$$

where $N$ is the overlapping number which is determined by the coherent length of the accelerationroll wave, $\delta$ is the phase difference after one orbital motion, $\omega$ is the angular speed of the sun rotation. The above equation is a multi-slit interference formula in optics, for a larger $N$ it is called as the Fabry-Perot interference formula.


Fig. 4 The head of the acceleration-roll wave may overlap with its tail.

The acceleration-roll wave function $\psi$ needs a further explanation. In quantum mechanics, $|\psi|^{2}$ equals to the probability of finding an electron due to Max Burn's explanation; in astrophysics, $|\psi|^{2}$ equals to the probability of finding a nucleon (proton or neutron) averagely in astronomic scale, because all mass is mainly made of nucleons, we have

$$
\begin{equation*}
|\psi|^{2} \propto \text { nucleon_density } \tag{20}
\end{equation*}
$$

It follows from the multi-slit interference formula that the interference intensity at maxima is proportional to $N^{2}$, that is

$$
\begin{equation*}
N^{2}=\frac{\left|\psi(0)_{\text {multi-wavelet }}\right|^{2}}{\left|\psi(0)_{\text {one-wavelet }}\right|^{2}} . \tag{21}
\end{equation*}
$$

What matter plays the role of "one-wavelet" in the solar core or Earth core? We choose vapor above the sea on the earth surface as the "reference matter: one-wavelet". Thus, the overlapping number $N$ is estimated by

Although today there is not vapor on the solar surface, the solar core has a maximum density of $1.5 \mathrm{e}+5 \mathrm{~kg} / \mathrm{m}^{3}$ [31], comparing to the vapor density $1.29 \mathrm{~kg} / \mathrm{m}^{3}$ on the earth, the solar overlapping number $N$ is estimated as $N=341$. The Earth core density is $5.53 \mathrm{e}+3 \mathrm{~kg} / \mathrm{m}^{3}$, the Earth's overlapping number $N$ is estimated as $N=65$.

For the Sun, Earth and Mars, their central densities and their reference matter density are given in the Table 2. Thus, their overlapping numbers are estimated also in this table.

Table 2 Estimating the overlapping number $N$ by comparing solid core to reference matter, regarding protons and neutrons as basis particles.

| object | Solid core, <br> density <br> $\left(\mathrm{kg} / \mathrm{m}^{3}\right)$ | Reference matter, <br> density $\left(\mathrm{kg} / \mathrm{m}^{3}\right)$ | Overlapping <br> number $N$ | $\beta$ <br> $\left(\mathrm{m} / \mathrm{s}^{2}\right)$ |
| :---: | :---: | :---: | :---: | :---: |
| Sun | $1.5 \mathrm{e}+5$ (max.) | 1.29 (vapor above the sea) | 341 | $2.961520 \mathrm{e}+10$ |
| Earth | 5530 | 1.29 (vapor above the sea) | 65 | $1.377075 \mathrm{e}+14$ |
| Mars | 3933.5 | 1.29 (vapor above the sea) | 55 | $2.581555 \mathrm{e}+15$ |
| Jupiter | 1326 |  |  | $4.016793 \mathrm{e}+13$ |
| Saturn | 687 |  |  | $7.183397 \mathrm{e}+13$ |
| Uranus | 1270 |  |  | $1.985382 \mathrm{e}+15$ |
| Neptune | 1638 |  | $2.077868 \mathrm{e}+15$ |  |
| Alien-planet | 5500 | 1.29 (has water on the surface) | 65 |  |

Sun's rotation angular speed at the equator is known as $\omega=2 \pi /(25.05 * 24 * 3600)$, unit s ${ }^{-1}$. Its mass $1.9891 \mathrm{e}+30(\mathrm{~kg})$, radius $6.95 \mathrm{e}+8(\mathrm{~m})$, mean density $1408\left(\mathrm{~kg} / \mathrm{m}^{3}\right)$, the solar core has a maximum density of $1.5 \mathrm{e}+5 \mathrm{~kg} / \mathrm{m}^{3}$ [31], the ultimate acceleration $\beta=2.961520 \mathrm{e}+10\left(\mathrm{~m} / \mathrm{s}^{2}\right)$, the constant $h M=9.100745 \mathrm{e}+14\left(\mathrm{~m}^{2} / \mathrm{s}\right)$. According to the $N=341$, the matter distribution of the $|\psi|^{2}$ is calculated out in Fig.5, it agrees well with the general description of the sun's interior. The radius of the Sun is calculated out to be $\mathrm{r}=7 \mathrm{e}+8(\mathrm{~m})$ with a relative error $0.72 \%$ in the Fig. 5 , it indicates that the sun radius strongly depends on the sun self-rotation.


Fig. 5 The matter distribution $|\psi| 2$ around the Sun has been calculated in radius direction.

[^0]
## 5. Earth central state and space debris distribution

Appling the acceleration-roll wav to the Moon, as illustrated in Fig.6(a), the Moon has been assigned a quantum number of $n=2$ in author's early study [28]. According to Eq.(15), the ultimate acceleration is fitted out to be $\beta=1.377075 \mathrm{e}+14\left(\mathrm{~m} / \mathrm{s}^{2}\right)$ in the Earth system. Another consideration is to take the quasi-satellite's perigee into account, for the moon and 2004_GU9 etc., as shown in Fig.6(b), but this consideration requires further understanding to its five quasi-satellites [28].


Fig. 6 orbital quantization for moon and quasi-satellites to the Earth, $H=h M$.
$<$ Clet2020 Script>//[26]
char $\operatorname{str}[200]$;int $\mathrm{i}, \mathrm{j}, \mathrm{k}, \mathrm{N}, \mathrm{nP}[10]$; double $\mathrm{x}, \mathrm{y}, \mathrm{z}, \mathrm{M}, \mathrm{r}_{\text {_ unit, a, }}$, B, B, H,r_ave[20], dP[10], D[1000];
double orbit $[10]=\{0,2.57,0$,$\} ; double \mathrm{e}[10]=\{0, \overline{0} .0549,0,0,0,0, \overline{0}, 0,0,0$,$\} ;$
int qn $[10]=\{0,2,3,4,5,6,7,8,9 ., 10$,$\} ;$
char Stars[100]=\{"Earth;Moon;"\};
int main( ) $\left\{\mathrm{N}=2 ; \mathrm{M}=5.97237 \mathrm{E} 24 ; \mathrm{r}_{2}\right.$ unit=1.495978707e8;
for $(\mathrm{i}=0 ; \mathrm{i}<\mathrm{N} ; \mathrm{i}+=1)\left\{\mathrm{x}=\operatorname{orbit}[\mathrm{i}] ; \mathrm{y}=\mathrm{e}[\mathrm{i}] ; \mathrm{z}=\mathrm{x} *\left(1+\mathrm{sqrt}\left(1-\mathrm{y}^{*} \mathrm{y}\right)\right) / 2 ; \mathrm{r}\right.$ ave $[\mathrm{i}]=\mathrm{z} ; / /$ average_radius
$D[i+i]=q n[i] ; D[i+i+1]=\operatorname{sqrt}(z) ;\}$
DataJob("REGRESSION,2",D,dP); $\mathrm{b}=\mathrm{dP}[0] ; \mathrm{a}=\mathrm{dP}[1]$;
SetAxis(X_AXIS, $0,0,3$, " $n ; 0 ; 1 ; 2 ; 3 ; "$ );
SetAxis(Y_AXIS, $0,0,3$, " \#if\#rsr\#t (average radius unit:0.001 AU); $0 ; 1 ; 2 ; 3 ;{ }^{-}$);
DrawFramé (0x0166,1,0xafffaf); Polyline(N,D);
SetPen(2,0xff0000); Plot("OVALFILL,0,2,XY,3,3,",D);
for $(\mathrm{i}=0 ; \mathrm{i}<\mathrm{N} ; \mathrm{i}+=1)\{\mathrm{nP}[0]=$ TAKE; $\mathrm{nP}[1]=\mathrm{i} ; \operatorname{TextJob}(\mathrm{nP}, S \operatorname{Stars}, \operatorname{str}) ; \mathrm{x}=\mathrm{qn}[\mathrm{i}]+0.2 ; \mathrm{y}=\operatorname{sqrt}(\operatorname{orbit}[\mathrm{i}])-0.05 ; \operatorname{TextHang}(\mathrm{x}, \mathrm{y}, 0, \operatorname{str}) ;\}$
$x=G R A V I T Y C * M^{*} r$ unit $; z=\operatorname{sqrt}(x) ; H=z^{*} a ; B=-z^{*} b ;$
TextAt(100,450,"\#if̄$\# t=\%$ e $\left.\# i f B \# t=\% e^{\prime \prime}, H, B\right)$;
for $(\mathrm{i}=0 ; \mathrm{i}<\mathrm{N} ; \mathrm{i}+=1)\{\mathrm{y}=\mathrm{b}+\mathrm{a} * \mathrm{qn}[\mathrm{i}] ; \mathrm{D}[\mathrm{i}+\mathrm{i}]=\mathrm{qn}[\mathrm{i}] ; \mathrm{D}[\mathrm{i}+\mathrm{i}+1]=\mathrm{y} ;\}$
SetPen(1,0x0000ff);Polyline(N,D,0.5,2.2,"quantization");//check
\} \} \#v07=?>A

Now let us talk about the central state of the earth, the earth's rotation angular speed is known as $\omega=2 \pi /(24 * 3600)$, unit s ${ }^{-1}$. Its mass $5.97237 \mathrm{e}+24(\mathrm{~kg})$, radius $6.371 \mathrm{e}+6(\mathrm{~m})$, core density $5530\left(\mathrm{~kg} / \mathrm{m}^{3}\right)$, the ultimate acceleration $\beta=1.377075 \mathrm{e}+14\left(\mathrm{~m} / \mathrm{s}^{2}\right)$, the constant $h M=1.956611 \mathrm{e}+11\left(\mathrm{~m}^{2} / \mathrm{s}\right)$.

We have estimated that the wave overlapping number in the central state of the earth is $N=65$, the matter distribution $|\psi|^{2}$ in radius direction is calculated out as shown in Fig.7(a), where the selfrotation near its equator has the period of 24 hours:

$$
\begin{equation*}
\delta(r)=\frac{1}{h M} \oint_{L}\left(v_{l}\right) d l=\frac{2 \pi r}{h M} \omega r=\frac{2 \pi \beta \omega r^{2}}{c^{3}} \tag{23}
\end{equation*}
$$

There is a central maximum of matter distribution at the earth heart, which gradually decreases to zero near the earth surface, then rises the secondary peaks and attenuates down off. The radius of the earth is calculated out to be $\mathrm{r}=6.4328 \mathrm{e}+6(\mathrm{~m})$ with a relative error $0.86 \%$ using the interference of its acceleration-roll wave. Space debris over the atmosphere has a complicated evolution [7,8], has itself speed

$$
\begin{equation*}
v_{l}=\sqrt{\frac{G M}{r}} ; \quad \delta(r)=\frac{1}{h M} \oint_{L}\left(v_{l}\right) d l=\frac{\beta}{c^{3}} \oint_{L}\left(v_{l}\right) d l=\frac{2 \pi \beta}{c^{3}} \sqrt{G M r} \tag{24}
\end{equation*}
$$

The secondary peaks over the atmosphere up to 2000km altitude is calculated out in Fig.7(b) which agrees well with the space debris observations [16]; the peak near 890 km altitude is due principally to the January 2007 intentional destruction of the Fengyun-1C weather spacecraft while the peak centered at approximately 770 km altitude was created by the February 2009 accidental collision of Iridium 33 (active) and Cosmos 2251 (derelict) communication spacecraft [16,18]. The observations based on the incoherent scattering radar EISCAT ESR located at $78^{\circ} \mathrm{N}$ in Jul. 2006 and in Oct. 2015 [21,22,23] are respectively shown in Fig.7(c) and (d). This prediction to secondary peaks also agrees well with other space debris observations [24,25].


Fig. 7 (a) The radius of the Earth is calculated out $\mathrm{r}=6.4328 \mathrm{e}+6(\mathrm{~m})$ with a relative error $0.86 \%$ by the interference of its acceleration-roll wave; (b) The prediction of the space debris distribution up to 2000 km altitude; (c) The pace debris distribution in Jul. 2006, Joint observation based on the incoherent scattering radar EISCAT ESR located at $78^{\circ} \mathrm{N}$ [21]; (d) The space debris distribution in Oct. 2015, Joint observation based on the incoherent scattering radar EISCAT ESR located at $78^{\circ} \mathrm{N}$ [21].

[^1]$<$ Clet2020 Script>//[26]
int i,j,k,m,n,N,nP[10]; double H,B,M,v r,r,AU,r unit, X,y,z,delta,D[10],S[10000];
double rs,rc,rot,a,b,atm_height,p,T,R1, $\overline{\mathrm{R}} 2, \mathrm{R} 3$; char $\operatorname{str}[100]$; int
Debris[96] $=\{110,0,237,0,287,0,317,2,320,1,357,5,380,1,387,4,420,2,440,3,454,14,474,9,497,45,507,26,527,19,557,17,597,34,63$
4,37,664,37,697,51,727,55,781,98,808,67,851,94,871,71,901,50,938,44,958,44,991,37,1028,21,1078,17,1148,10,1202,9,1225,6, $1268,12,1302,9,1325,5,1395,7,1395,18,1415,36,1429,12,1469,22,1499,19,1529,9,1559,5,1656,4,1779,1,1976,1$,
$\operatorname{main}()\{\mathrm{k}=80 ; \mathrm{rs}=6.378 \mathrm{e} 6 ; \mathrm{rc}=0 ; \mathrm{atm}$ height $=1.5 \mathrm{e} 5 ; \mathrm{n}=0 ; \mathrm{N}=65$;
$\mathrm{H}=1.956611 \mathrm{e} 11 ; \mathrm{M}=5.97237 \mathrm{e} 24 ; \mathrm{A} \overline{\mathrm{U}}=1.496 \mathrm{E} 11 ; \mathrm{r}$ _unit $=1 \mathrm{e} 4$;
rot $=2 * \mathrm{PI} /(24 * 60 * 60) ; / /$ angular speed of the Earth
$\mathrm{b}=\mathrm{PI} /\left(2 * \mathrm{PI}^{*} \mathrm{rot} * \mathrm{rs} * \mathrm{rs} / \mathrm{H}\right) ; \mathrm{R} 1=\mathrm{rs} / \mathrm{r}$ _unit;R2=(rs+atm_height)/r unit;R3=(rs+2e6)/r_unit;
for( $\mathrm{i}=\mathrm{R} 2 ; \mathrm{i}<\mathrm{R} 3 ; \mathrm{i}+=1)\left\{\mathrm{r}=\mathrm{abs}(\mathrm{i}) * \mathrm{r}\right.$ - unnit; delta=2*PI*sqrt(GRAV$\left.I T Y C * \mathrm{M}^{*} \mathrm{r}\right) / \mathrm{H}$;
$y=S u m J o b(" S L I T A D D, @ N, @$ delta",D); $y=1 \mathrm{e} 3 * \mathrm{y} /(\mathrm{N} * \mathrm{~N})$;// visualization scale:1000
if $(\mathrm{y}>1) \mathrm{y}=1 ; \mathrm{S}[\mathrm{n}]=\mathrm{i} ; \mathrm{S}[\mathrm{n}+1]=\mathrm{y} ; \mathrm{n}+=2 ;\}$
SetAxis(X AXIS,R1,R1,R3,"altitude; r\#sds\#t;500;1000;1500;2000km ;");
SetAxis(Y_AXIS, $0,0,1$,"\#if $|\psi| \#$ su2\#t; $0 ; ~ ; 1 \mathrm{e}-3 ; ") ;$ DrawFrame(FRAME_SCALE, 1,0xafffaf); $\mathrm{x}=\mathrm{R} 1+(\mathrm{R} 3-\mathrm{R} 1) / 5$;
SetPen(1,0xff0000);Polyline(n/2,S,x,0.8,"\#ifly|\#su2\#t (density, prediction)");
for $(\mathrm{i}=0 ; \mathrm{i}<48 ; \mathrm{i}+=1)\{\mathrm{S}[\mathrm{i}+\mathrm{i}]=\mathrm{R} 1+(\mathrm{R} 3-\mathrm{R} 1) * \operatorname{Debris}[\mathrm{i}+\mathrm{i}] / 2000 ; \mathrm{S}[\mathrm{i}+\mathrm{i}+1]=\operatorname{Debris}[1+\mathrm{i}+1] / 300 ;\}$
SetPen(1,0x0000ff);Polyline(48,S,x,0.7,"Space debris (2018, observation) ");
\} $\# \mathrm{v} 07=$ ? $>\mathrm{A}$

## 6. Mars and Jovian planets

The Mars and its satellites are quantized very well by its ultimate acceleration $\beta$ as shown in Fig.8(a). Now let us talk about the Mars in the central state with quantum number $n=0$, its ultimate acceleration is $\beta=2.581555 \mathrm{e}+15\left(\mathrm{~m} / \mathrm{s}^{2}\right)$ in the Mars system. We have estimated the Mars overlapping number $N=55$ in Table 2, the matter distribution $|\psi|^{2}$ around the Mars can been calculated out in radius direction as shown in Fig.8(b), where the self-rotation at equator has the period of 24 hours.

The radius of the Mars is calculated out as $\mathrm{r}=1.6 \mathrm{e}+6(\mathrm{~m})$ with a relative error $52.79 \%$, no further attempt was made to improve the calculation, because of limited knowledge about the Mars history. But one thing is certain, the Mars has frequently bombarded with smaller objects, in fact, consequently with the inclination of 25.2 degrees, so that its self-rotation is unstable or incorrect for its formation in a sense. At very beginning, the Mars self-rotation should have a period of 100 hours. Thanks to the Mars for safeguarding the Earth.


Fig. 8 (a)Quantization of Mars and its satellites. (b)The matter distribution $|\psi|^{2}$ around the Mars has been calculated in radius direction.

```
<Clet2020 Script>//[26]
int i,j,k,m,n,N,nP[10];
double H,B,M,v_r,r,AU,r_unit,x,y,z,delta,D[10],S[1000],F[1000];
double rs,rc,rot,a,b,atm height,beta; char str[100];
main() {k=80;rs=3.389e\overline{e};\textrm{rc}=0;\textrm{atm}\mathrm{ height =10e3;n=0; N=55;}
beta=2.581555e+15;H=SPEEDC*SPEEDC*SPEEDC/beta;
M=0.107*5.97237e24; AU=1.496E11;r_unit=1e5;
rot=2*PI/(24*60*60);//angular speed
for(i=-k;i<k;i+=1) {r=abs(i)*r_unit;
if(r
delta=2*PI* v_r/H;y=\overline{SumJob("SLIT AMDD,@N,@delta",D); y=y/(N*N);}
if(y>1) y=1; \[n]=1;S[n+1]=y; if(i>0 &&& rc==0 && y<0.001)rc=r; n+=2;}
SetAxis(X_AXIS,-k,0,k,"r; ;;;");SetAxis(Y_AXIS,0,0,1.2,"#if|\psi|#su2#t;0;0.4;0.8;1.2;");
DrawFrame(FRAME_SCALE,1,0xafffaf); x=50;z=100*(rs-rc)/rs;
SetPen(1,0xff0000);Polyline(k+k,S,k/3,1," nucleon_density");
r=rs/r_unit;y=-0.05;D[0]=-r;D[1]=y;D[2]=r;D[3]=y;
SetPen(2,0x0000ff); Draw("ARROW,3,2,XY,10,100,10,10,",D);
```

In order to extend the quantization rule to the Jovian planets (Jupiter, Saturn, Uranus and Neptune), it is necessary to further study the magnet-like components of gravity [28], the issue beyond the scope of this paper.

## 7. New aspect: coherent width and sunspot cycle

The coherence length of waves is usually mentioned but the coherence width of waves is rarely discussed in quantum mechanics, simply because the latter is not a matter for electrons, nucleon or photos, but it is a matter in astrophysics. Analyzing observational data tells us that the coherent width of an acceleration-roll wave can extend to a 1000 km or even more in planetary scale, as illustrated in Fig.9(a), the overlapping can even happen in width-direction bringing with a new aspect for wave interference.

In the solar convective zone, adjacent convective arrays form a top-layer flow, a middle-layer gas and a ground-layer reverse flow, like the concept of molecular current in electromagnetism. Considering one convective ring at the equator as shown in Fig.9(b), there is an apparent velocity difference between the top-layer flow and the middle-layer gas, where their acceleration-roll waves are denoted respectively by
(a)

(b)


Fig. 9 (a) Illustration of overlapping in coherent width direction. (b) Convective rings at the equator, the speed difference causes a beat frequency

\footnotetext{
<Clet2020 Script>// [26]
int i, j, k, R,D[500];
main()\{DrawFrame(FRAME NULL, 1,0xafffaf);
$\mathrm{R}=60$; SetPen(1,0xffff00);
D[0]=-R; D[1]=-R; D[2]=R; D[3]=R; Draw("ELLIPSE, 1,2,XY,0",D);
R=85; k=15; SetPen(1,0xff0000);
D $[0]=-\mathrm{R} ; \mathrm{D}[1]=-\mathrm{R} ; \mathrm{D}[2]=\mathrm{R} ; \mathrm{D}[3]=\mathrm{R} ;$ Draw("ELLIPSE, $0,2, \mathrm{XY}, 0 ", \mathrm{D}) ;$
$\mathrm{D}[0]=0 ; \mathrm{D}[1]=0 ; \mathrm{D}[2]=\mathrm{R}-\mathrm{k} ; \mathrm{D}[3]=0 ; \mathrm{D}[4]=\mathrm{R}+\mathrm{k} ; \mathrm{D}[5]=0$;
Draw("SECTOR, 1,3,XY,15,30,130,0",D);
R=95; k=15; SetPen(1,0x00ff);
D $[0]=-R ; D[1]=-R ; D[2]=R ; D[3]=R ;$ Draw("ELLIPSE, $0,2, X Y, 0 ", D)$;
$\mathrm{D}[0]=0 ; \mathrm{D}[1]=0 ; \mathrm{D}[2]=\mathrm{R}-\mathrm{k} ; \mathrm{D}[3]=0 ; \mathrm{D}[4]=\mathrm{R}+\mathrm{k} ; \mathrm{D}[5]=0$;
Draw("SECTOR, 1,3,XY,15,0,100,0",D);D[4]=R+k+k;
Draw("SECTOR,3,3,XY, 15,0,100,0",D);
TextHang( $0,0,0$,"Solid core"); TextHang(R-k-k,-k, 0 , "Coherent width");
TextHang( $0, \mathrm{R}+\mathrm{k}+\mathrm{k}+\mathrm{k}, 0$, "Coherent length");TextHang(-R,R+k,0,"Overlapping"); \}\#v07=?>A

```
<Clet2020 Script>//Clet is a C compiler[26]
double beta,H,M,N,dP[20],D[2000],r,rs,rot,x,y,v1,v2,K1,K2,T1,T2,T,Lamda,V; int i,j,k;
int main() {beta=2.961520e10; H=SPEEDC*SPEEDC*SPEEDC/beta;
M=1.9891E30; rs=6.95e8;rot=2*PI/(25.05*24*3600);v1=rot*rs;K1=v1*v1/2;//T1=2*PI*H/K1;
v2=0.7346*sqrt(BOLTZMANN*5700/MP) +0.2485*sqrt(BOLTZMANN*5700/(MP+MP));
K2=v2*v2/2;T2=2*PI*H/(K2-K1);T=T2/24*3600*365.2422;
Lamda=2*PI*H/(v2-v1);V=Lamda/T2;
SetViewAngle("temp0,theta60,phi-60");
DrawFrame(FRAME_LINE,1,0xafffaf);Overlook("2,1,60", D);
```

}
$/ / \operatorname{TextAt}(10,10, " v 1=\% d, v 2=\% \mathrm{~d}, \mathrm{~T}=\% .2 \mathrm{f} y, \lambda=\% \mathrm{e}, \mathrm{V}=\% \mathrm{~d} ", \mathrm{v} 1, \mathrm{v} 2, \mathrm{~T}$, Lamda,V);
SetPen(1,0x4f4fff); for $(\mathrm{i}=0 ; \mathrm{i}<18 ; \mathrm{i}+=1)\{\mathrm{v} 1=\mathrm{i} * 2 * \mathrm{PI} / 18 ; \mathrm{x}=70 * \cos (\mathrm{v} 1) ; \mathrm{y}=70 * \sin (\mathrm{v} 1) ; \operatorname{Ring}() ;\}$
SetPen(2,0xff0000); Draw("ARROW,0,2,XYZ,15","80,0,0,80,60,0");
TextHang(100,20,0,"top-layer $\omega \#$ sd2\#t"); SetPen(2,0x0000ff);
Draw("ARROW,0,2,XYZ,15","70,0,0,70,60,0");
TextHang(50,60,0,"center $\omega \#$ sd1\#t");TextHang(140,-30,0," $\omega \#$ sdbeat\#t= $\omega \# s d 2 \# t-\omega \#$ sd1 1 t" $)$;
Ring() $\{\mathrm{k}=0 ; \mathrm{N}=20 ; \mathrm{r}=10$;
for $(\mathrm{j}=0 ; \mathrm{j}<\mathrm{N}+2 ; \mathrm{j}+=1)\{\mathrm{k}=\mathrm{j}+\mathrm{j}+\mathrm{j} ; \mathrm{v} 2=j * 2 * \mathrm{PI} / \mathrm{N}$;
$\mathrm{D}[\mathrm{k}]=\mathrm{x}+\mathrm{r} * \cos (\mathrm{v} 2) ; \mathrm{D}[\mathrm{k}+1]=\mathrm{y}+\mathrm{r} * \sin (\mathrm{v} 2) ; \mathrm{D}[\mathrm{k}+2]=0 ;\}$
Plot("POLYLINE,4,22,XYZ,8",D);\}
\#v07=?>A

$$
\begin{align*}
& \psi=\psi_{\text {middle }}+C \psi_{\text {top }} \\
& \psi_{\text {middle }}=\exp \left[\frac{i \beta}{c^{3}} \int_{L}\left(v_{1} d l+\frac{-c^{2}}{\sqrt{1-v_{1}^{2} / c^{2}}} d t\right)\right]  \tag{25}\\
& \psi_{\text {top }}=\exp \left[\frac{i \beta}{c^{3}} \int_{L}\left(v_{2} d l+\frac{-c^{2}}{\sqrt{1-v_{2}^{2} / c^{2}}} d t\right)\right]
\end{align*}
$$

Where, we simply suppose the couple coefficient to be $C=1$, their overlapping wave interference in coherent width direction leads to a beat phenomenon

$$
\begin{align*}
& |\psi|^{2}=\left|\psi_{\text {middle }}+\psi_{\text {top }}\right|^{2}=2+2 \cos \left[\frac{2 \pi}{\lambda_{\text {beat }}} \int_{L} d l-\frac{2 \pi}{T_{\text {beat }}} t\right] \\
& \frac{2 \pi}{T_{\text {beat }}}=\frac{\beta}{c^{3}}\left(\frac{c^{2}}{\sqrt{1-v_{2}^{2} / c^{2}}}-\frac{c^{2}}{\sqrt{1-v_{1}^{2} / c^{2}}}\right) \simeq \frac{\beta}{c^{3}}\left(\frac{v_{2}^{2}}{2}-\frac{v_{1}^{2}}{2}\right)  \tag{26}\\
& \frac{2 \pi}{\lambda_{\text {beat }}}=\frac{\beta}{c^{3}}\left(v_{2}-v_{1}\right) ; \quad V=\frac{\lambda_{\text {beat }}}{T_{\text {beat }}}=\frac{1}{2}\left(v_{2}+v_{1}\right)
\end{align*}
$$

Their speeds are calculated by

$$
\begin{align*}
& v_{1}=\omega r_{\text {middle }}=2017(\mathrm{~m} / \mathrm{s}) \quad(\text { sun rotation })  \tag{27}\\
& v_{2} \approx 6200(\mathrm{~m} / \mathrm{s}) \quad(\approx \text { observed in Evershed flow })
\end{align*}
$$

There are three ways to estimate the top-layer flow speed $v_{2}$. (1) regarding the Evershed flow as the eruption of the top-layer flow, about $6 \mathrm{~km} / \mathrm{s}$ speed was reported [31]. (2) regarding the prominences as the eruption of the top-layer flow; the prominence turbulent speeds are reported [31] in the range $2-10 \mathrm{~km} / \mathrm{s}$. (3) Alternatively, since the thermal equilibrium gas in a convective zone supplies the flow speed $v_{2}$, where the temperature $T=5700^{\circ} \mathrm{K}$, the flow consists of $73.46 \%$ hydrogen atoms and $24.85 \%$ helium atoms; these atoms are approximately regarded as in 1 D circular flow: $m v^{2} / 2=k T / 2$. Thus, the top-layer speed $v_{2}$ can be estimated out by

$$
\begin{align*}
& v_{2} \approx 0.7346 \sqrt{\frac{k T}{m_{\text {hydrogen }}}}+0.2485 \sqrt{\frac{k T}{m_{\text {helium }}}} .  \tag{28}\\
& =6244(\mathrm{~m} / \mathrm{s})
\end{align*}
$$

Their beat period $T_{\text {beat }}$ is calculated out to be a very remarkable value 10.38 (years), in agreement with the sunspot cycle value (say, mean 11years).

$$
\begin{equation*}
T_{b e a t} \simeq \frac{4 \pi c^{3}}{\beta\left(v_{2}^{2}-v_{1}^{2}\right)}=10.38(\text { years }) \tag{29}
\end{equation*}
$$

The relative error to the mean 11 years is $5.6 \%$ for the beat period calculation using the accelerationroll waves. This beat turns out to be a nucleon density up and down that undergoes to drive the sunspot cycle evolution. The beat wavelength $\lambda_{\text {beat }}$ is too long to observe, only the beat period is easy to be observed. This nucleon density up and down is understood as a new type nuclear reaction in astronomic scale.

In the above calculation, although this seems to be a rough model, there is an obvious correlation between solar radius, solar rotation, solar density, ultimate acceleration and Planck-constant-like constant $h$.

## 8. The El Niño Oscillation in the Earth ocean

Now we discuss the beat phenomenon between the oceans and the atmospheric circulation near the equator via width-direction interference to their acceleration-roll waves.

The trade winds or easterlies are the permanent east-to-west prevailing winds that flow in the Earth's equatorial region. The trade winds blow mainly from the northeast in the Northern Hemisphere and from the southeast in the Southern Hemisphere, strengthening during the winter and when the Arctic oscillation is in its warm phase. Trade winds have been used by captains of sailing ships to cross the world's oceans for centuries. The driving force of atmospheric circulation is the uneven distribution of solar heating across the earth, which is greatest near the equator and least at the poles. This air rises to the tropopause, about $10-15$ kilometers above sea level, where the air is no longer buoyant [33].


Fig. 10 Illustration of overlapping in coherent width direction.
<Clet2020 Script>// [26]
double beta,H,M,N,r,rc, rs,rot, v1, v2,K1,K2,T1,T2,T,Lamda, V;
int i, j, k, R,D[50];
int main() \{beta $=1.377075 \mathrm{e}+14 ; \mathrm{H}=$ SPEEDC $* S P E E D C * S P E E D C / b e t a ;$
$\mathrm{M}=5.97237 \mathrm{e} 24 ; \mathrm{rs}=6.371 \mathrm{e} 6 ; \mathrm{rot}=2 * \mathrm{PI} /(24 * 3600) ; \mathrm{v} 1=\mathrm{rot}^{*} \mathrm{rs} ; \mathrm{K} 1=\mathrm{v} 1 * \mathrm{v} 1 / 2 ; / / \mathrm{T} 1=2 * \mathrm{PI} * \mathrm{H} / \mathrm{K} 1$;
v2=v1-10;//sqrt(M*GRAVITYC/rs);
$\mathrm{K} 2=\mathrm{v} 2 * \mathrm{v} 2 / 2 ; \mathrm{T} 2=2 * \mathrm{PI}^{*} \mathrm{H} /(\mathrm{K} 2-\mathrm{K} 1) ; \mathrm{T}=\mathrm{T} 2 / 24 * 3600 * 365.2422$;
Lamda $=2 * \mathrm{PI}^{*} \mathrm{H} /(\mathrm{v} 2-\mathrm{v} 1) ; \mathrm{V}=\mathrm{Lamda} / \mathrm{T} 2$;
DrawFrame(FRAME NULL,1,0xafffaf)
$/ / \operatorname{TextAt}(10,10, " \mathrm{v} 1=\% \mathrm{~d}, \mathrm{v} 2=\% \mathrm{~d}, \mathrm{~T}=\% .2 \mathrm{f} y, \lambda=\% \mathrm{e}, \mathrm{V}=\% \mathrm{~d} \mathrm{~m}, \mathrm{v} 1, \mathrm{v} 2, \mathrm{~T}$, Lamda,V);
$\mathrm{R}=80$; SetPen(1,0xffff00); D[0]=-R; D[1]=-R; D[2]=R; D[3]=R; Draw("ELLIPSE, 1,2,XY,0",D);
SetPen(2,0xff0000); Draw("ELLIPSE,0,2,XY,0",D);Draw("ARROW,0,2,XY,8","80,0,80,50");
$R=85 ; \quad \operatorname{Set} \operatorname{Pen}(2,0 x f f 00 f f) ; D[0]=-R ; D[1]=-R ; D[2]=R ; D[3]=R ;$
Draw("ELLIPSE,0,2,XY,0",D);Draw("ARROW,0,2,XY, 8","85,0,85,50");
TextHang(-30,0,0,"Solid core\#nEquator plan"); TextHang(70,70,0,"\#ifv\#sd1\#se v\#sd2\#t"); \} \#v07=? $>\mathrm{A}$

As a simplest model, regarding the ocean as a static massive matter with the earth self-rotation speed $v_{1}=\omega r=463 \mathrm{~m} / \mathrm{s}$ at the equator, approximately estimating the mean speed of the atmospheric circulation at the equator as $\mathrm{v}_{2}=\mathrm{v}_{1}-10 \mathrm{~m} / \mathrm{s}$, (magnitude order estimation), then they give out their
beat period as

$$
\begin{equation*}
T_{b e a t} \simeq \frac{4 \pi c^{3}}{\beta\left(v_{2}^{2}-v_{1}^{2}\right)}=8.5(\text { years }) \tag{30}
\end{equation*}
$$

As shown in Fig.10, this beat turns out to be a nucleon density up and down that undergoes to drive the cycle evolution. This period 8.5 years of the nucleon density up and down on the earth surface is also a remarkable cycle. It is easy to find that the ocean variation near the equator seems having a clear positive response to this cycle. El Niño-Southern Oscillation (ENSO) is an irregular periodic variation (4 years-10 years) in winds and sea surface temperatures over the tropical eastern Pacific Ocean, affecting the climate of much of the tropics and subtropics. The warming phase of the sea temperature is known as El Niño and the cooling phase as La Niña. The Southern Oscillation is the accompanying atmospheric component, coupled with the sea temperature change: El Niño is accompanied by high air surface pressure in the tropical western Pacific and La Niña with low air surface pressure there. The two periods last several months each and typically occur every few years with varying intensity per period [33].

## 9. Atmosphere

Above the sea level, air pressure and density decrease with altitude in the atmosphere. The temperature has a more complicated profile with altitude, and may remain relatively constant or even increase with altitude in some regions, as shown in Fig.11(a). Earth's atmosphere can be divided into five main layers: troposphere ( 0 to 12 km ), stratosphere ( 12 to 50 km ), mesosphere ( 50 to 80 km ), thermosphere ( 80 to 700 km ), and exosphere ( 700 to $10,000 \mathrm{~km}$ ) [34].

The Earth molecules move under the control of their acceleration-roll wave whose coherent length is so long that the wave has to overlap as

$$
\begin{align*}
& \psi(r)=1+e^{i \delta}+e^{i 2 \delta}+\ldots+e^{i(N-1) \delta}=\frac{1-\exp (i N \delta)}{1-\exp (i \delta)} \\
& \delta(r)=\frac{\beta}{c^{3}} \oint_{L}\left(v_{l}\right) d l=\frac{\beta 2 \pi \omega r^{2}}{c^{3}} \tag{31}
\end{align*}
$$

The theoretical Earth radius $r_{c}$ should appear at the first minimal density that is

$$
\begin{equation*}
N \delta=2 \pi \quad \Rightarrow \quad N \frac{\beta 2 \pi \omega r_{c}^{2}}{c^{3}}=2 \pi \tag{32}
\end{equation*}
$$

Let $r_{\mathrm{s}}$ denote the Earth radius at sea level, the atmosphere height is defined as $h=r_{c}-r_{s}$, then we calculate out these parameters from the above equation at the equator as: $r_{s}=6.378000 \mathrm{e}+6(\mathrm{~m})$, $r_{c}=6.433725 \mathrm{e}+6(\mathrm{~m}), h=5.572515 \mathrm{e}+4(\mathrm{~m})$. Substituting into the density Eq. 31 ), using the core density $5530\left(\mathrm{~kg} / \mathrm{m}^{3}\right)$, we calculate atmospheric density at the sea level to be

$$
\begin{align*}
& \rho_{0}=5530\left(\mathrm{~kg} / \mathrm{m}^{3}\right) \\
& \rho_{r=r s}=\rho_{0}\left|\psi_{r=r s}\right|^{2}=1.7\left(\mathrm{~kg} / \mathrm{m}^{3}\right) \tag{33}
\end{align*}
$$

In a sense, this result agrees well with the real air-vapor density $1.29\left(\mathrm{~kg} / \mathrm{m}^{3}\right)$ at the sea level, this satisfaction indicates that Eq.(31) is a platform qualified for atmosphere study. The atmospheric height is literally $h=55.7 \mathrm{~km}$, within the range $r<r_{c}$ in where Earth self-rotation supports the gas
speed $v=\omega r$; while in the rang $r>r_{c}$ the gravity supports space debris speed $v=\operatorname{sqrt}(G M r)$ without Earth self-rotation, so that there is a density sudden-change near $r_{c}$, as shown in Fig.11(b). In other words, the atmosphere lives in the rang $r_{s}<r<r_{c}$; the space debris lives in the rang $r>r_{c}$.


Fig. 11 (a) Atmosphere density (blue line) and temperatures (red line) against altitude [32]. (b) Predicted atmosphere density (red line) and real measurement (blue line).
<Clet2020 Script> //[26]
double $D \mathrm{D}[38]=\{1.31,0.00,1.20,0.53,1.10,1.32,1.01,2.11,0.90,3.16,0.81,4.21,0.70,5.53,0.60,6.58,0.51,8.16,0.40,9.74,0.31$, $11.84,0.2 \overline{0}, 15.00,0.14,17.11,0.10,19.47,0.07,21.84,0.04,25.26,0.02,27.89,0.01,31.84,0.00,50.00$, double
D $\mathrm{T}[20]=\{1.37,0.00,0.68,10.79,0.68,20.00,0.80,32.63,1.21,48.42,1.20,52.11,1.08,60.00,0.41,80.26,0.40,90.26,0.52,99.74$, main()
SetAxis(X AXIS, $0,0,1.5$, "D(kg/m\#su3\#t);0;0.5;1;1.5;")
SetAxis(XA AXIS, $0,0,1.5$, "T(K);150;200;250;300;");
SetAxis(Y_AXIS,0,0,100,"Altitude(km);0;10;20;30;40;50;60;70;80;90;100;")
DrawFrame(FRAME BOX,1,0xafffaf);
SetPen(2,0x0000ff); Polyline(19,D D)
SetPen(2,0xff0000); Polyline (10,D_T);
TextHang(0.2,20,0,"Density - Temperature");
\# $\mathrm{v} 007=$ ? $>\mathrm{A}$
<Clet2020 Script>//[26]
int i,j,k,m,n,N,nP[10]; double H,B,M,v r,r,AU,r unit, x,y,z,delta,D[10],S[10000]
double rs,rc,rot,a,b,atm height,p,T,R1,R2,R3; char str[100];
double $\mathrm{D} \quad \mathrm{D}[38]=\{1.31,0.00,1.20,0.53,1.10,1.32,1.01,2.11,0.90,3.16,0.81,4.21,0.70,5.53,0.60,6.58,0.51,8.16,0.40,9.74,0.31$, $11.84,0.2 \overline{0}, 15.00,0.14,17.11,0.10,19.47,0.07,21.84,0.04,25.26,0.02,27.89,0.01,31.84,0.00,50.00$, $\operatorname{main}()\{\mathrm{k}=80 ; \mathrm{rs}=6.378 \mathrm{e} 6 ; \mathrm{n}=0 ; \mathrm{N}=65$;
$\mathrm{H}=1.956611 \mathrm{e} 11 ; \mathrm{M}=5.97237 \mathrm{e} 24 ; \mathrm{AU}=1.496 \mathrm{E} 11 ; \mathrm{r}$ unit=1e3;
rot $=2 * \mathrm{PI} /(24 * 60 * 60)$;//angular speed of the Earth
$\mathrm{a}=\mathrm{N} * \mathrm{PI} * \mathrm{rot} / \mathrm{H} ; \mathrm{rc}=\mathrm{sqrt}(\mathrm{PI} / \mathrm{a}) ; \mathrm{atm}$ height=rc-rs;
$\mathrm{b}=\mathrm{PI} /\left(2 * \mathrm{PI} * \mathrm{rot}^{*} \mathrm{rs} * \mathrm{rs} / \mathrm{H}\right) ; \mathrm{R} 1=\mathrm{rs} / \mathrm{r} \_u n i t ; \mathrm{R} 2=\mathrm{R} 1 ; \mathrm{R} 3=(\mathrm{rs}+3 \mathrm{e} 5) / \mathrm{r} \_$unit;
for (i=R2;i<R3;i+=1) \{r=i*r_unit;
if(r<rs+atm height) delta=2*PI* rot*r*r/H;else delta=2*PI*sqrt(GRAVITYC*M*r)/H;
y=SumJob("SLIT ADD,@N,@delta",D); y/=(N*N); y*=5530;S[n]=i;S[n+1]=y;n+=2;
SetAxis(X AXIS, R1,R1,R3,"altitude; r\#sds\#t;150km;300km ;");
SetAxis(Y_AXIS, $0,0,2$,"\#ifp\#tt(kg/m\#su3\#t);0;1;2;");DrawFrame(FRAME_SCALE, 1,0xafffaf);
$\mathrm{x}=\mathrm{R} 1+(\mathrm{R} 3-\mathrm{R} 1) / 5$; SetPen(2,0xff0000) ;Polyline(n/2,S,x,1.8,"\#ifp\#ie(density, prediction)");
for $(\mathrm{i}=0 ; \mathrm{i}<19 ; \mathrm{i}+=1)\left\{\mathrm{S}[\mathrm{i}+\mathrm{i}]=\left(\mathrm{rs}+\mathrm{D} \mathrm{D}[\mathrm{i}+\mathrm{i}+1]^{*} 1000\right) / \mathrm{r}\right.$ unit; $\left.\mathrm{S}[\mathrm{i}+\mathrm{i}+1]=\mathrm{D} \mathrm{D}[\mathrm{i}+\mathrm{i}] ;\right\}$
SetPen(2,0x0000ff) ;Polyline(19,S,x,1.5,"Measurement")
delta=2*PI* rot*rs*rs/H; y=SumJob("SLIT_ADD,@N,@delta",D); y=5530*y/(N*N);
/Format(str,"rc=\%e $\quad \mathrm{h}=\% \mathrm{e} \quad$ density=$\% \mathrm{f}^{1}$ ",rc, atm_height, y); TextAt(10,30,str); TextHang(rc/r_unit,-0.1,0,"r\#sdc\#t");
\#v07=?>A

According to this quantum gravity theory, the atmospheric height $h=55.7 \mathrm{~km}$ within the range $r_{s}<r<r_{c}$ includes troposphere and stratosphere. The acceleration-roll waves can also be used to explain atmospheric circulation which is the motion of the air through the troposphere in large-scale, and to explain ocean circulation [28].

## 10. Global climate change

Detailed estimates of temperature and carbon dioxide levels over the past 800000 years can be made from Antarctic ice samples. Layers of ice have accumulated for more than a million years there. Drilling down through the ice layers, scientists can examine bubbles of gas to find ancient $\mathrm{CO}_{2}$ levels, and examine the percentage of different oxygen isotopes to estimate the temperature. These detailed measurements are shown in Fig.12. Within this time range, Earth's temperature has gone up and down, with glaciation and $\mathrm{CO}_{2}$ levels [32].


Fig. 12 Earth's temperature has gone up and down, with $\mathrm{CO}_{2}$ levels.

Milankovitch cycles describe the collective effects of changes in the Earth's movements on its climate over thousands of years. The term was coined and named after Serbian geophysicist and astronomer Milutin Milankovitch. In the 1920s, he hypothesized that variations in eccentricity, axial tilt, and precession combined to result in cyclical variations in the intra-annual and latitudinal distribution of solar radiation at the Earth's surface, and that this orbital forcing strongly influenced the Earth's climatic patterns. Similar astronomical hypotheses had been advanced in the 19th century by Joseph Adhemar, James Croll, and others, but verification was difficult because there was no reliably dated evidence in that time, and because it was unclear which periods were important. In recent years, materials on Earth that have been unchanged for millennia (obtained via ice, rock, and deep ocean sediment cores) have been studied to indicate the history of Earth's climate. Although broadly consistent with the Milankovitch hypothesis, a number of specific observations are not explained by the hypothesis.

The Earth self-rotation $\omega$ represent the angular speed of the Earth solid core, the angular speed $\omega_{\text {air }}$ of the atmosphere may differ from the core $\omega$ a little bit due to $\mathrm{CO}_{2}$ impurity. Considering the constancy of rotational inertia of the air layer, the difference is estimated by

$$
\begin{align*}
& I=m \omega r^{2} \Rightarrow \frac{\Delta \omega_{a i r}}{\omega_{a i r}}=-\frac{\Delta \rho_{a i r}}{\rho_{a i r}} \approx-\frac{\Delta \rho_{C O 2}}{\rho_{a i r}}  \tag{34}\\
& \Delta \rho_{C O 2}=\rho_{C O 2}(t)-\rho_{i d l e}
\end{align*}
$$

where $\rho_{\text {idle }}$ denotes the amount of $\mathrm{CO}_{2}$ which did not contribute to the rotation variation $\Delta \omega$ because its impulse has been absorbed by the Earth solid core for a long time, the $\rho_{\text {idle }}$ is about 40 ppm in this calculation. The overlapping interference takes place between the air acceleration-roll wave and the ground acceleration-roll wave within their coherent width, as shown in Fig.13(a), they are

$$
\begin{align*}
& \psi=C \psi_{\text {ground }}+\psi_{\text {air }} \\
& \psi_{\text {ground }}=\exp \left[\frac{i \beta}{c^{3}} \int_{L}\left(v_{1} d l+\frac{-c^{2}}{\sqrt{1-v_{1}^{2} / c^{2}}} d t\right)\right]  \tag{35}\\
& \psi_{\text {air }}=\exp \left[\frac{i \beta}{c^{3}} \int_{L}\left(v_{2} d l+\frac{-c^{2}}{\sqrt{1-v_{2}^{2} / c^{2}}} d t\right)\right]
\end{align*}
$$

Where $C$ is the couple coefficient, their overlapping interference in coherent width direction leads to a beat phenomenon at the radius $r$ location

$$
\begin{align*}
& T_{b e a t} \simeq \frac{4 \pi c^{3}}{\beta\left(v_{2}^{2}-v_{1}^{2}\right)}=100604(\text { years })  \tag{36}\\
& v_{1}=\omega r ; \quad v_{2}=\left(\omega+\Delta \omega_{\text {air }}\right) r
\end{align*}
$$

Thus, the $\mathrm{CO}_{2}$ density is given by

$$
\begin{equation*}
\rho_{\mathrm{CO}_{2}}^{2}=\rho_{\text {mean_CO2 }^{2}}+C^{2} \rho_{\text {mean_CO2 }^{2}}+2 C \rho_{\text {mean_CO2 }^{2}}{ }^{2} \cos \left(\frac{2 \pi}{T_{\text {beat }}} t+\varphi_{0}\right) \tag{37}
\end{equation*}
$$

Over the past 800000 years, the mean $\mathrm{CO}_{2}$ density in the air is about 220 ppm , the couple coefficient $C=0.136$, then it is easy to calculate out the beat period of $\mathrm{CO}_{2}$ density to be the remarkable 100.6 kiloyears, this result agrees well with the Antarctic ice measurements, as shown in Fig.13(b). The beat wavelength $\lambda_{\text {beat }}$ is too long to observe, only the beat period is easy to be observed.


Fig. 13 Earth's temperature has gone up and down, with $\mathrm{CO}_{2}$ levels.

[^2]$<$ Clet2020 Script>// [26]
double
D_T $[426]=\{800.00,255.80,795.30,256.38,784.73,315.28,782.38,307.70,777.68,304.20,772.98,298.37,771.81,287.29,769.46,292$.
$54,765.93,274.46,762.41,283.79,758.88,266.30,755.36,280.29,750.66,256.97,740.09,254.64,735.39,291.37,730.69,289.04,728.3$ $4,280.87,725.99,274.46,723.64,282.04,720.12,256.38,715.42,275.63,713.07,282.62,710.72,290.20,706.02,293.12,697.80,306.53$, $693.10,306.53,691.92,308.86,681.35,280.87,680.18,264.55,676.65,281.46,675.48,273.29,674.30,277.38,670.78,262.22,668.43,2$ $66.88,666.08,254.64,666.08,251.72,662.56,262.22,659.03,257.55,654.33,268.05,651.98,261.05,648.46,265.71,647.28,261.63,64$ $3.76,261.05,639.06,254.05,636.71,256.38,633.19,252.89,632.01,255.80,628.49,257.55,624.96,301.28,621.44,300.12,619.09,304$ $78,617.91,304.78,614.39,311.20,612.04,309.45,608.52,311.20,606.17,301.87,603.82,289.04,600.29,282.62,596.77,270.38,595.5$ $9,286.71,589.72,276.79,589.72,291.95,582.67,268.05,576.80,310.03,567.40,307.70,565.05,303.62,559.18,270.96,556.83,295.45$, $552.13,270.96,552.13,277.96,548.60,266.88,546.26,258.13,542.73,266.30,540.38,262.22,536.86,268.63,534.51,262.80,528.63,2$ $87.87,527.46,284.96,521.59,293.12,518.06,292.54,513.36,277.96,512.19,276.21,509.84,286.12,506.31,281.46,505.14,288.45,49$ $3.39,293.12,489.87,306.53,488.69,293.12,480.47,276.79,480.47,288.45,473.42,268.05,474.60,288.45,471.07,256.97,464.02,265$. $71,460.50,252.89,456.98,269.80,453.45,262.22,448.75,272.13,449.93,264.55,441.70,256.97,439.35,250.55,435.83,256.97,427.6$ $1,318.19,426.43,304.78,420.56,325.77,417.03,321.69,409.99,341.52,406.46,335.10,400.59,317.03,393.54,272.71,390.01,294.87$, $386.49,277.38,384.14,288.45,378.27,260.47,377.09,277.96,371.22,260.47,371.22,266.30,367.69,261.05,366.52,272.13,360.65,2$ $52.89,353.60,268.63,345.37,251.14,337.15,348.51,334.80,322.27,330.10,330.44,324.23,284.96,319.53,308.86,312.48,266.88,31$ $0.13,287.29,308.96,268.63,303.08,280.87,301.91,262.22,298.38,266.30,294.86,290.20,287.81,275.63,281.94,256.38,276.06,248$. $80,273.72,257.55,267.84,273.29,265.49,262.80,263.14,273.29,256.09,257.55,251.40,274.46,246.70,335.69,242.00,304.20,234.9$ $5,259.30,230.25,278.54,226.73,256.38,219.68,314.11,214.98,304.20,213.80,311.78,209.10,295.45,207.93,284.96,205.58,307.70$, $198.53,266.88,197.36,284.37,193.83,260.47,191.48,290.20,189.13,257.55,187.96,252.89,183.26,261.05,180.91,275.63,176.21,2$ $60.47,175.04,275.63,173.86,264.55,172.69,273.29,167.99,254.05,165.64,273.88,165.64,261.05,160.94,273.29,158.59,249.97,15$ $3.89,268.63,150.37,256.97,148.02,263.97,143.32,254.64,140.97,257.55,131.57,356.68,129.22,329.85,124.52,337.43,119.82,329$. $85,112.78,282.04,111.60,265.71,108.08,297.20,106.90,269.80,103.38,293.12,99.85,281.46,95.15,293.12,89.28,271.55,83.41,297$. $78,81.06,274.46,76.36,293.70,77.53,273.29,71.66,281.46,72.83,262.22,65.79,248.80,58.74,283.21,56.39,264.55,52.86,283.79,48$ $.16,265.13,46.99,290.79,44.64,256.97,39.94,272.13,35.24,253.47,32.89,268.05,29.37,247.64,24.67,256.38,18.80,249.97,12.92,2$ 98.37,9.40,325.77,5.87,310.61,2.35,326.36,1.17,318.19,1,320,\};
double
D_CO2 $[338]=\{797.65,190.50,794.13,188.75,783.55,260.47,783.55,249.97,775.33,245.31,771.81,231.31,768.28,237.73,765.93,2$ $15.57,762.41,224.90,758.88,207.99,757.71,217.32,750.66,194.58,748.31,182.33,740.09,179.42,736.56,210.90,731.86,205.66,72$ $1.29,213.82,717.77,188.75,713.07,220.82,709.54,219.65,702.50,233.06,698.97,227.81,694.27,235.98,688.40,238.31,683.70,220$. $23,677.83,229.56,669.60,193.41,664.90,175.92,661.38,191.08,657.86,187.00,655.51,198.66,650.81,186.41,647.28,193.99,643.7$ $6,189.91,643.76,193.99,639.06,186.41,636.71,193.99,633.19,192.83,624.96,236.56,616.74,253.47,609.69,259.30,602.64,238.89$, $596.77,219.65,593.25,237.14,590.90,230.15,588.55,249.39,583.85,209.15,576.80,251.14,573.27,249.39,565.05,254.05,559.18,2$ $29.56,555.65,249.39,553.30,223.73,547.43,205.07,543.91,209.74,540.38,203.32,536.86,210.32,533.33,192.24,527.46,221.98,51$ $8.06,245.89,512.19,240.06,511.01,236.56,509.84,240.64,502.79,230.15,495.74,238.31,489.87,248.80,486.34,232.48,480.47,220$. $82,479.30,238.31,475.77,233.64,473.42,242.97,469.90,203.91,466.37,208.57,459.32,192.83,455.80,205.66,454.63,193.99,446.4$ $0,205.66,440.53,198.66,438.18,199.83,424.08,272.71,424.08,268.05,419.38,277.96,415.86,275.04,411.16,284.96,406.46,277.96$, $404.11,282.04,400.59,287.29,397.06,276.79,392.36,252.30,387.67,263.97,379.44,228.40,374.74,227.81,371.22,202.16,366.52,2$ $06.82,364.17,188.16,352.42,219.65,345.37,204.49,335.98,297.20,333.63,268.63,333.63,273.88,323.05,252.89,323.05,235.39,32$ $0.70,269.80,314.83,240.64,313.66,227.81,310.13,248.22,304.26,237.14,301.91,217.32,298.38,207.41,294.86,232.48,290.16,229$. $56,280.76,195.74,277.24,188.75,271.37,203.91,272.54,204.49,267.84,227.23,266.67,209.15,259.62,196.33,253.74,212.65,246.7$ $0,278.54,242.00,243.56,239.65,250.55,233.77,228.98,227.90,235.98,227.90,205.07,218.50,252.30,214.98,244.14,214.98,256.97$, $209.10,230.73,203.23,249.97,198.53,223.15,190.31,231.31,185.61,203.32,182.09,216.73,178.56,191.66,172.69,198.66,167.99,1$ $88.16,164.46,204.49,159.77,188.75,155.07,201.57,145.67,191.66,142.14,196.91,130.40,286.71,130.40,270.96,122.17,277.96,12$ $1.00,268.05,115.12,275.04,108.08,233.64,103.38,237.73,99.85,228.98,96.33,230.73,89.28,213.24,85.76,240.64,82.23,221.40,74$. $01,229.56,66.96,193.41,59.91,221.40,52.86,192.83,49.34,214.99,46.99,190.50,42.29,211.49,38.77,211.49,29.37,189.33,27.02,19$ $2.24,23.49,185.25,10.57,265.13,9.40,259.88,3.52,268.63,-1.17,380.58$,
double rs, rot,beta, $\mathrm{H}, \mathrm{v} 1, \mathrm{v} 2$, phase, C , gamma, kiloyears, $\mathrm{r} 1, \mathrm{r} 2$;
double $\mathrm{x} 0, \mathrm{y} 0, \mathrm{x}, \mathrm{y}, \mathrm{x} 1, \mathrm{y} 1, \mathrm{x} 2, \mathrm{y} 2, \mathrm{a}, \mathrm{b}, \mathrm{D}[1000]$; int $\mathrm{i}, \mathrm{j}$;
main() $\{\mathrm{rs}=6.378 \mathrm{e} 6 ;$ beta $=1.377075 \mathrm{e}+14 ; \mathrm{H}=\mathrm{SPEEDC} * S P E E D C * S P E E D C / b e t a ;$ rot $=2 * \mathrm{PI} /(24 * 60 * 60) ; / /$ angular speed of the Earth
SetAxis(X_AXIS,-800,-800,0,"t\#n(kiloyears);-800;-600;-400;-200;0;");
SetAxis(Y-AXIS,180,180,380,"\#Y-10CO\#sd2\#t(ppm);180;230;380;330;380;");
SetAxis(YA AXIS,-14,0,8,"T( $\left.\left.{ }^{\circ} \mathrm{C}\right) ; ;-12 ;-8 ;-4 ; 0 ; 4 ; 8 ; "\right)$;
DrawFrame(FRAME_BOX,1,0xafffaf); $\mathbf{j}=213$;
for $(\mathrm{i}=0 ; \mathrm{i}<\mathrm{j} ; \mathrm{i}+=1)\left\{\mathrm{D}[\mathrm{i}+\mathrm{i}]=-\mathrm{D} \_\mathrm{T}[\mathrm{i}+\mathrm{i}] ; \mathrm{D}[\mathrm{i}+\mathrm{i}+1]=\mathrm{D} \_\mathrm{T}[\mathrm{i}+\mathrm{i}+1] ;\right\}$
SetPen(2,0x0000ff); Polyline(j, D,-750,360,"Temperature");j=169;
for $(\mathrm{i}=0 ; \mathrm{i}<\mathrm{j} ; \mathrm{i}+=1)\left\{\mathrm{D}[\mathrm{i}+\mathrm{i}]=-\mathrm{D}\right.$ CO2 $\left.[\mathrm{i}+\mathrm{i}] ; \mathrm{D}[\mathrm{i}+\mathrm{i}+1]=\mathrm{D} \_\mathrm{CO} 2[\mathrm{i}+\mathrm{i}+1] ;\right\}$
SetPen(2,0x00ffff); Polyline(可,D,-750,340,"CO\#sd2\#t");
kiloyears $=1000 * 24 * 3600 * 365.2422 ; x 0=-800 *$ kiloyears; $x 1=x 0 ; \operatorname{Predict} 2() ;\}$
Predict1 () $\left\{y=1.8 \mathrm{e}-6 ; \mathrm{v} 1=\right.$ rot $^{*} \mathrm{rs} ; y 0=0.4 \mathrm{e}-6$;
$\mathrm{v} 2=\mathrm{v} 1 *(1-\mathrm{y}) ; \mathrm{a}=(\mathrm{v} 1 * \mathrm{v} 1-\mathrm{v} 2 * \mathrm{v} 2) /(2 * \mathrm{H}) ; \mathrm{r} 1=2 * \mathrm{PI} /(\mathrm{a}$ *kiloyears $) ; \mathrm{r} 2=0$;
$\mathrm{C}=30 / 220 ; \mathrm{y} 1=220 ; \mathrm{y} 2=\mathrm{C}^{*} \mathrm{y} 1 ; \mathrm{j}=169 ;$ phase $=-\mathrm{PI} / 4$;
for $(\mathrm{i}=0 ; \mathrm{i}<\mathrm{j} ; \mathrm{i}+=1)\left\{\mathrm{x}=\mathrm{D} \_\right.$CO2 $[\mathrm{i}+\mathrm{i}] ; \mathrm{x}=-\mathrm{x} *$ kiloyears; $\mathrm{y}=\mathrm{D} \_\mathrm{CO} 2[\mathrm{i}+\mathrm{i}+1] ; \mathrm{y}^{*}=1 \mathrm{e}-8 ; \mathrm{r} 2+=\mathrm{y}$;
$\mathrm{v} 2=\mathrm{v} 1 *(1-\mathrm{y}+\mathrm{y} 0) ; \quad \mathrm{a}=(\overline{\mathrm{v}} 1 * \mathrm{v} 1-\mathrm{v} 2 * \mathrm{v} 2) /(2 * \mathrm{H}) ;$ phase $+=\mathrm{a}$ 芷 $(\mathrm{x}-\mathrm{x} 1)$;
$\mathrm{a}=\mathrm{y} 1 * \mathrm{y} 1+\mathrm{y} 2 * \mathrm{y} 2+2 * \mathrm{y} 1 * \mathrm{y} 2 * \cos ($ phase $) ; \mathrm{b}=\operatorname{sqrt}(\mathrm{a}) ; \quad \mathrm{x} 1=\mathrm{x}$;
$\mathrm{D}[\mathrm{i}+\mathrm{i}]=-\mathrm{D} \operatorname{CO} 2[\mathrm{i}+\mathrm{i}] ; \mathrm{D}[\mathrm{i}+\mathrm{i}+1]=\mathrm{b} ;\}$
SetPen(2,0xff0000); Polyline(i,D,-750,320,"Prediction");
TextAt $10,10,{ }^{\prime} \mathrm{v} 1=\% \mathrm{f}, \mathrm{v} 2=\% \mathrm{f}, \# \mathrm{nrl}=\% \mathrm{e}, \mathrm{r} 2=\% \mathrm{e}$ ", v1,v2, $\left.\left.\mathrm{r} 1, \mathrm{r} 2 / \mathrm{j}\right) ;\right\}$
Predict2 ()$\{j=169$; gamma $=13.16 ; y 0=2.26 \mathrm{e}-6$;
for $(\mathrm{i}=0 ; \mathrm{i}<\mathrm{j} ; \mathrm{i}+=1)\left\{\mathrm{x}=\mathrm{D}\right.$ CO2 $[\mathrm{i}+\mathrm{i}] ; \mathrm{x}=-\mathrm{x} *$ kiloyears; $\mathrm{y}=\mathrm{D} \quad \mathrm{CO} 2[\mathrm{i}+\mathrm{i}+1] ; \mathrm{y}^{*}=1 \mathrm{e}-8$;
$a=-5+273 *($ gamma*44/28.966-1)*0.00417*(y-y0)/y0; $\bar{b}=330+a * 200 / 22$;
$\left.\mathrm{D}[\mathrm{i}+\mathrm{i}]=-\mathrm{D} \_\mathrm{CO} 2[\mathrm{i}+\mathrm{i}] ; \mathrm{D}[\mathrm{i}+\mathrm{i}+1]=\mathrm{b} ;\right\}$
SetPen(2,-0xff0000); Polyline(i,D,-750,320, "Prediction");
\} \#v07=? $>\mathrm{A}$

The global warming mainly arises from $\mathrm{CO}_{2}$ impurity. The atmosphere pressure is given by

$$
\begin{equation*}
p=\int_{0}^{\infty}\left(\rho_{\text {air }}+\Delta \rho\right) g d h=\int_{0}^{\infty}\left(1+\frac{\Delta \rho}{\rho_{\text {air }}}\right) \rho_{\text {air }} g d h \tag{38}
\end{equation*}
$$

Because the ratio of $\Delta \rho$ and $\rho_{\text {air }}$ is not a constant, the above integral cannot carry out generally. At a higher altitude, the air density goes to lower, the ratio goes to larger. Let $n_{\text {air }}$ denotes the air
molecular density whose mean molecular mass is 28.966 ; $n_{\mathrm{CO} 2}$ denotes the $\mathrm{CO}_{2}$ molecular density whose molecular mass is 44 . The pressure variation due to the impurity increasement is estimated by

$$
\begin{equation*}
\frac{\Delta p}{p}=\gamma \frac{\Delta \rho}{\rho_{\text {air }}}=\gamma\left(\frac{44}{28.966}\right) \frac{\Delta n_{C O 2}}{n_{\text {air }}} . \tag{39}
\end{equation*}
$$

The $\gamma$ constant is named as stack coefficient for the impurity, it will be determined by experiments. The atmosphere temperature is given by

$$
\begin{align*}
& p=n k T \\
& \frac{\Delta T}{T}=\frac{\Delta p}{p}-\frac{\Delta n}{n_{\text {air }}}=\left(\gamma \frac{44}{28.966}-1\right)\left(\frac{n_{C O 2}}{n_{\text {air }}}\right)\left(\frac{\Delta n_{C O 2}}{n_{C O 2}}\right) \tag{40}
\end{align*}
$$

By mole fraction (i.e., by number of molecules), dry air contains $78.08 \%$ nitrogen, $20.95 \%$ oxygen, $0.93 \%$ argon, $0.0417 \%$ carbon dioxide, and small amounts of other gases. Air also contains a variable amount of water vapor, on average around $1 \%$ at sea level, and $0.4 \%$ over the entire atmosphere. After fitting $\mathrm{CO}_{2}$ density over last 800000 years, the stack coefficient takes $\gamma=13.16$. Therefore, the global warming is predicted by the above equation as shown in Fig.14, the result agrees well with the observation from ice samples.


Fig. 14 Earth's temperature has gone up and down, with $\mathrm{CO}_{2}$ levels.

Levels of carbon dioxide have risen by about $30 \%$ over the last century, as shown in Fig. 15. This increase far exceeds any natural changes over the last 10 million years, so most scientists attribute it to the increased burning of fossil fuels by humans. Human activities such as deforestation, mining coal and metal elements, have also reduced the places where carbon can be stored. Over the same time period, average ocean and air temperatures have increased by about $0.8^{\circ} \mathrm{C}$ [32], a phenomenon referred to as global warming.

Over the past century, human activities --- various mining, extracting fossil fuels into the atmosphere, airplanes flying in the sky, etc., they all cause a tiny change in the atmosphere. The stack coefficient $\gamma$ also takes $\gamma=13.16$, this number is obtained from the past 800000 years data. Thus, the global warming is predicted by the above equation as shown in Fig. 15, the global temperature has increased about $0.8^{\circ} \mathrm{C}$ for the past century, this result agrees well with the estimation $0.8^{\circ} \mathrm{C} /$ last-
century in textbook [32].


Fig. 15 The prediction of global warming agrees well with the observation for the past century.

## <Clet2020 Script> //[26]

double
D_T[196] $=\{1880.00,314.94,1881.61,320.36,1882.69,318.19,1883.77,303.01,1884.57,309.52,1887.00,298.40,1888.61,325.24,18$ 89.69,305.45,1890.76,302.20,1891.84,305.72,1892.65,304.64,1896.41,326.05,1897.76,317.65,1900.18,330.12,1900.72,330.12,1 903.14,308.70,1905.56,324.16,1906.91,308.70,1907.44,316.02,1909.06,314.40,1909.60,318.46,1911.21,316.30,1912.29,317.65, $1913.63,329.85,1914.44,330.66,1916.86,306.81,1917.94,310.33,1918.74,324.43,1919.55,321.17,1920.90,324.70,1921.97,320.63$ ,1922.24,318.73,1923.59,321.99,1924.66,321.45,1926.01,331.75,1926.28,326.33,1927.62,327.95,1928.70,318.19,1930.31,330.3 $9,1931.93,329.58,1932.74,322.80,1933.81,330.93,1934.89,323.61,1936.23,335.54,1937.85,337.44,1938.65,329.31,1939.46,336$. $90,1941.61,334.73,1942.42,331.75,1943.23,334.19,1944.30,329.31,1945.65,329.85,1946.19,335.27,1949.42,320.63,1952.11,335$ .27,1953.45,324.97,1954.53,325.51,1955.61,320.09,1956.41,333.37,
$1957.49,334.46,1959.64,329.31,1960.72,333.92,1962.06,331.20,1963.68,316.30,1966.64,328.77,1967.71,324.97,1969.60,331.20$ ,1970.67,324.70,1971.75,326.87,1972.56,339.61,1973.63,326.87,1974.71,327.41,1975.78,317.65,1976.32,337.71,1977.67,334.1 $9,1980.63,350.99,1981.43,334.46,1982.24,347.20,1983.59,337.71,1984.39,336.63,1987.35,352.35,1988.70,345.84,1989.51,355$. 60,1990.31,353.70,1991.93,337.98,1994.62,354.25,1995.70,350.18,1996.77,351.81,1997.85,368.89,1998.65,355.06,1999.46,353 16,2001.61,368.61,2003.23,365.09,2004.57,373.77,2005.38,367.53,2006.46,371.05,2007.53,363.19,2008.61,370.51,\}; double
D_CO2[30] $=\{1880.00,290.00,1894.80,293.25,1918.74,299.22,1943.50,307.08,1955.61,312.23,1967.17,320.63,1974.17,326.60,1$ 980.90,333.92,1987.62,341.23,1993.00,348.28,1997.31,356.14,2001.08,363.73,2004.30,370.51,2006.46,376.20,2009.15,382.44, \};
double k,y0, x,y,gamma,D[500],a,b;int i,j
main() \{
SetAxis(X AXIS,1880,1880,2040,"Year;1880;1920;1960;2000;2040;");
SetAxis(Y AXIS,280,280,400,"\#Y-10CO\#sd2\#t(ppm);280;300;320;340;360;380;400;");
SetAxis(YĀ_AXIS,-1,0,1,"T( $\left.\left.{ }^{\circ} \mathrm{C}\right) ; ~ ;-0.4 ; 0 ; 0.4 ; 0.8 ; 1.2 ; "\right) ;$
DrawFrame_(FRAME_BOX,1,0xafffaf);j=98;
SetPen(2,0x0000ff); $\overline{\text { Polyline(j,D_T, 1900,365,"Temperature"); }}=15$;
SetPen(2,0x00ffff); Polyline(j,D_CO2,1900,380,"CO\#sd2\#t");
$\mathrm{j}=15$; y0=3e-6; gamma=13.16;
for $(\mathrm{i}=0 ; \mathrm{i}<\mathrm{j} ; \mathrm{i}+=1)\left\{\mathrm{x}=\mathrm{D}\right.$ CO2 $[\mathrm{i}+\mathrm{i}] ; \mathrm{y}=\mathrm{D}$ CO2 $[\mathrm{i}+\mathrm{i}+1] ; \mathrm{y}^{*}=1 \mathrm{e}-8$;
$\mathrm{a}=-0.2+273 *($ gamma $* \overline{4} 4 / 28.966-1) * 0 . \overline{0} 0417^{*}(\mathrm{y}-\mathrm{y} 0) / \mathrm{y} 0 ; \mathrm{b}=320+\mathrm{a} * 200 / 22$;
$\mathrm{D}[\mathrm{i}+\mathrm{i}]=\mathrm{x} ; \mathrm{D}[\mathrm{i}+\mathrm{i}+1]=\mathrm{b} ;\}$
SetPen(2,0xff0000); Polyline(j,D,1900,350,"Prediction");
\} $\# \mathrm{v} 07=$ ? $>\mathrm{A}$

## 11. Conclusions

In analogy with the ultimate speed $c$, there is an ultimate acceleration $\beta$, nobody's acceleration can exceed this limit $\beta$, in the solar system, $\beta=2.961520 \mathrm{e}+10\left(\mathrm{~m} / \mathrm{s}^{2}\right)$. Because this ultimate acceleration is a large number, any effect connecting to $\beta$ will become easy to test, including quantum gravity tests. In this paper, an approach is put forward to connect the ultimate acceleration with quantum theory, as an application, this relativistic model gives out the sunspot cycle to be 10.38 years due to the ultimate acceleration. The same approach is applied to the Earth El Niño oscillation and global climate change cycle, calculating out the El Niño irregular period to be 8.5 years; calculating out the global climate change cycle to be 100.6 kiloyears which coincides with the Milankovitch cycle; the global temperature has increased about $0.8^{\circ} \mathrm{C}$ for the past century.

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[^0]:    $<$ Clet2020 Script $>/ /[26]$
    int $\mathrm{i}, \mathrm{j}, \mathrm{k}, \mathrm{m}, \mathrm{n}, \mathrm{N}, \mathrm{nP}[10]$;
    double beta,H,B,M,r,r_unit,x,y,z,delta,D[1000],S[1000], a,b,rs,re,rot,atm height; char str[100];
    $\operatorname{main}()\{\mathrm{k}=150 ; \mathrm{rs}=6.95 \mathrm{e} 8 ; \mathrm{rc}=0 ; \mathrm{x}=25.05 ; \mathrm{rot}=2 * \mathrm{PI} /(\mathrm{x} * 24 * 3600) ; \mathrm{n}=0 ; \mathrm{N}=3 \overline{4} 1$;
    beta $=2.961520 \mathrm{e} 10 ; \mathrm{H}=\mathrm{SPEEDC} * \mathrm{SPEEDC} * \mathrm{SPEEDC} / \mathrm{beta} ; \mathrm{M}=1.9891 \mathrm{E} 30 ;$ atm_height=2e6; r_unit=1E7; $\mathrm{b}=\mathrm{PI} /\left(2 * \mathrm{PI}^{*}\right.$ rot*rs*rs/H);
    for $(\mathrm{i}=-\mathrm{k} ; \mathrm{i}<\mathrm{k} ; \mathrm{i}+=1)\left\{\mathrm{r}=\mathrm{abs}(\mathrm{i})^{*} \mathrm{r} \_\right.$unit;
    if( $\mathrm{r}<\mathrm{rs}+$ atm height $)$ delta $=2 * \overline{\mathrm{PI}} * \operatorname{rot}^{*} \mathrm{r}^{*} \mathrm{r} / \mathrm{H}$; else delta $=2 * \mathrm{PI} * \operatorname{sqrt}(\mathrm{GRAVITYC} * \mathrm{M} * \mathrm{r}) / \mathrm{H} ; / /$ around the star
    y=SumJob("SLIT_ADD,@N,@delta",D); y=y/(N*N);
    $\mathrm{S}[\mathrm{n}]=\mathrm{i} ; \mathrm{S}[\mathrm{n}+1]=\mathrm{y} ;$; $\mathrm{if}(\mathrm{i}>0 \& \& \mathrm{rc}==0 \& \& \mathrm{y}<0.001) \mathrm{rc}=\mathrm{r} ; \mathrm{D}[\mathrm{n}]=\mathrm{i} ; \mathrm{D}[\mathrm{n}+1]=\mathrm{z} ; \mathrm{n}+=2 ;\}$
    SetAxis(X_AXIS,-k,0,k,"\#ifr; ; ; ;");SetAxis(Y_AXIS,0,0,1.2,"\#if| $\mid$ |\#su2\#t; $0 ; 0.4 ; 0.8 ; 1.2 ; ")$;
    DrawFramé(FRAME_SCALE,1,0xafffaf); $z=1 \overline{0} 0 *(r s-r c) / r s ;$
    SetPen(1,0xff0000); Polyline(k+k,S,k/2,1," nucleon_density"); SetPen(1,0x0000ff); //Polyline(k+k,D);
    //Draw("LINE,0,2,XY,0","20,0.5,60,0.6");TextHang(60,0.6,0,"core");
    $\mathrm{r}=\mathrm{rs} / \mathrm{r}$ unit $; \mathrm{y}=-0.05 ; \mathrm{D}[0]=-\mathrm{r} ; \mathrm{D}[1]=\mathrm{y} ; \mathrm{D}[2]=\mathrm{r} ; \mathrm{D}[3]=\mathrm{y} ; \mathrm{Draw}(" A R R O W, 3,2, \mathrm{XY}, 10,100,10,10, ", \mathrm{D})$;
    Format(str,"\#ifN\#t=\%d\#n\#ifß\#t=\%e\#nrc=0e\#nrs=\%e\#nerror=\%.2f\%",N,beta,rc,rs,z);
    TextHang(k/2,0.7,0,str);TextHang(r+5,y/2,0,"\#ifr\#sds\#t");TextHang(-r,y+y,0,"Sun diameter"); \}\#v07=? $>\mathrm{A}$

[^1]:    <Clet2020 Script>//[26]
    int i,j,k,m,n,N,nP[10];
    double $\mathrm{H}, \mathrm{B}, \mathrm{M}, \mathrm{v} \_$r,r,AU, r _unit, $\mathrm{x}, \mathrm{y}, \mathrm{z}$, delta,D[10],S[1000];
    double rs,rc,rot,a,b,atm height,beta; char str[100];
    $\operatorname{main}()\{\mathrm{k}=80 ; \mathrm{rs}=6.378 \overline{\mathrm{e}} ; ; \mathrm{rc}=0 ; \mathrm{atm}$ height $=1.5 \mathrm{e} 5 ; \mathrm{n}=0 ; \mathrm{N}=65$;
    beta $=1.377075 \mathrm{e}+14 ; \mathrm{H}=$ SPEEDC*SPEEDC*SPEEDC/beta;
    $\mathrm{M}=5.97237 \mathrm{e} 24 ; \mathrm{AU}=1.496 \mathrm{E} 11 ; \mathrm{r}$ _unit $=1 \mathrm{e}-6^{*} \mathrm{AU}$;
    rot $=2 * \mathrm{PI} /(24 * 60 * 60) ; / /$ angular speed of the Earth
    for (i=-k;i<k;i+=1) \{r=abs(i)*r_unit;
    if( $\mathrm{r}<\mathrm{rs}+$ atm height $) \mathrm{v}$ _r $\mathrm{r}=\mathrm{rot}^{*} \mathrm{r}_{\mathrm{r}} \mathrm{F}_{\mathrm{r}}$; else $\mathrm{v} \quad \mathrm{r}=$ sqrt(GRAVITYC* $\mathrm{M}^{*} \mathrm{r}$ );//around the Earth
    delta=2*PI* v r/H; y=SumJob("SLIT ADDD,@N,@delta",D); y=y/(N*N);
    $\operatorname{if}(\mathrm{y}>1) \mathrm{y}=1 ; \mathrm{S}[\mathrm{n}]=\mathrm{i} ; \mathrm{S}[\mathrm{n}+1]=\mathrm{y} ; \mathrm{if}(\mathrm{i}>0 \quad \overline{\&} \& \mathrm{rc}==0 \& \& \mathrm{y}<0.001) \mathrm{rc}=\mathrm{r} ; \quad \mathrm{n}+=2 ;\}$
    SetAxis(X_AXIS,-k,0,k,"r; ; ; ;");SetAxis(Y_AXIS,0,0,1.2,"\#iflu|\#su2\#t;0;0.4;0.8;1.2;");
    DrawFrame (FRAME SCALE, $1,0 x a f f f a f) ; ~ x=50 ; z=100 *(r s-r c) / \mathrm{rs} ;$
    SetPen(1,0xff0000);Polyline(k+k,S,k/2,1," nucleon_density");
    $\mathrm{r}=\mathrm{rs} / \mathrm{r}$ unit $; \mathrm{y}=-0.05 ; \mathrm{D}[0]=-\mathrm{r} ; \mathrm{D}[1]=\mathrm{y} ; \mathrm{D}[2]=\mathrm{r} ; \mathrm{D}[3]=\mathrm{y}$;
    SetPen(2,0x0000ff); Draw("ARROW,3,2,XY,10,100,10,10,",D);
    Format(str,"\#ifN\#t=\%d\#n\#ifß\#t=\%e\#nrc=\%e\#nrs=\%e\#nerror=\%.2f\%",N,beta,rc,rs,z); TextHang(k/2,0.7,0,str);TextHang(r+5,y/2,0,"r\#sds\#t");TextHang(-r,y+y,0,"Earth diameter"); \} \#v07=?>A

[^2]:    <Clet2020 Script>// [26]
    double beta,H,M,N,r,rc, rs, a, y, rot,v1,v2,K1,K2,T1,T2,Lamda, V;
    int i, j, k, R,D[50];
    int main() \{beta=1.377075e+14; H=SPEEDC*SPEEDC*SPEEDC/beta;
    $\mathrm{M}=5.97237 \mathrm{e} 24 ; \mathrm{rs}=6.378 \mathrm{e} 6 ;$ rot $=2 * \mathrm{PI} /(24 * 3600) ; \mathrm{v}=\mathrm{rot} * \mathrm{rs} ; \mathrm{y}=1.8 \mathrm{e}-6$;
    $\mathrm{v} 2=\mathrm{v} 1 *(1-\mathrm{y}) ; \mathrm{a}=(\mathrm{v} 1 * \mathrm{v} 1-\mathrm{v} 2 * \mathrm{v} 2) /(2 * \mathrm{H}) ; \mathrm{T} 1=2 * \mathrm{PI} / \mathrm{a} ; \mathrm{T} 2=24 * 3600 * 365.2422$;
    Lamda=2*PI*H/(v2-v1);V=Lamda/T2;
    DrawFrame(FRAME NULL, 1,0xafffaf);
    $/ / T e x t A t(10,10, " v 1=\% \mathrm{~d}, \mathrm{v} 2=\% \mathrm{~d}, \mathrm{~T}=\% .2 \mathrm{f} y, \lambda=\% \mathrm{e}, \# \mathrm{nV}=\% \mathrm{~d} \mathrm{l}, \mathrm{v} 1, \mathrm{v} 2, \mathrm{~T} 1 / \mathrm{T} 2$, Lamda, V$)$;
    R=80; SetPen(1,0xffff00); D[0]=-R; D[1]=-R; D[2]=R; D[3]=R; Draw("ELLIPSE, 1,2,XY,0",D);
    SetPen(2,0xff0000); Draw("ELLIPSE, 0,2,XY,0",D);Draw("ARROW,0,2,XY,8","80,0,80,50");
    R=85; SetPen(2,0xff00ff); D[0]=-R; D[1]=-R; D[2]=R; D[3]=R;
    Draw("ELLIPSE, 0,2,XY,0",D);Draw("ARROW,0,2,XY,8","85,0,85,50");
    TextHang(-30,0,0,"Solid core\#nEquator plan"); TextHang(70,70,0,"\#ifv\#sd1\#se v\#sd2\#t");
    \} \#v07=?>A

