Gravitational Micro-Thrusters
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Here we show how to produce thrusts of the order of 100$kN$ or more, starting from sets of micro-tubes (diameter $<< 1$cm) filled with air at low pressure, subjected to gravity $g$, and a strong magnetic field $H$. Under these conditions, these micro-tubes work as micro-thrusters, where the thrust is produced starting from the local potential gravitational energy.

**Key words:** Gravitational Mass, Gravitational Interaction, Gravitational Thruster.

**INTRODUCTION**

In this paper we will show that micro-tubes (diameter $<< 1$cm) filled with air at low pressure, subjected to gravity $g$, and a strong magnetic field $H$, can work as micro-thrusters, where the thrust is produced starting from the local potential gravitational energy. In this context, it is also shown that sets of these micro-thrusters can produce thrust of the order of 100$kN$ or more.

**THEORY**

In a previous paper [1] we shown that there is a correlation between the gravitational mass, $m_g$, and the rest inertial mass $m_{i0}$, which is given by

$$
\chi = \frac{m_g}{m_{i0}} = 1 - 2 \left[ \left( \frac{\Delta p}{m_0 c} \right)^2 - 1 \right] = 1 - 2 \left[ \frac{\left( \frac{U_{n_r}}{m_0 c^2} \right)^2}{1 + \left( \frac{W_{n_r}}{\rho c^2} \right)^2} - 1 \right] = 1 - 2 \left[ \frac{\left( \frac{\varepsilon_1}{m_0} c^2 \right)^2}{1 + \left( \frac{\varepsilon_2}{\mu_0} c^2 \right)^2} - 1 \right]
$$

(1)

where $\Delta p$ is the variation in the particle’s kinetic momentum; $U$ is the electromagnetic energy absorbed or emitted by the particle; $n_r$ is the index of refraction of the particle; $W$ is the density of energy on the particle ($F/kg$); $\rho$ is the matter density ($kg/m^3$) and $c$ is the speed of light.

The instantaneous values of the density of electromagnetic energy in an electromagnetic field can be deduced from Maxwell’s equations and has the following expression

$$
W = \frac{1}{2} \varepsilon E^2 + \frac{1}{2} \mu H^2
$$

(2)

where $E = E_m \sin \omega t$ and $H = H \sin \omega t$ are the instantaneous values of the electric field and the magnetic field respectively.

It is known that $B = \mu H$, $E = \omega / k_r$ [2] and

$$
v = \frac{dz}{dt} = \frac{\omega}{\kappa_r} = \frac{c}{\sqrt{\varepsilon_0 \mu_0 \left[ 1 + (c \omega) \right]^2}}
$$

(3)

where $k_r$ is the real part of the propagation vector $\vec{k}$ (also called phase constant); $k = \vec{k} = k_r + ik_i$; $\varepsilon$, $\mu$ and $\sigma$ are the electromagnetic characteristics of the medium in which the incident (or emitted) radiation is propagating ($\varepsilon = \varepsilon_0$; $\varepsilon_0 = 8.854 \times 10^{-12} F/m$; $\mu = \mu_0 \mu_0$ where $\mu_0 = 4\pi \times 10^{-7} H/m$; $\sigma$ is the electrical conductivity in $S/m$). From Eq. (3), we see that the index of refraction $n_r = c/v$ is given by

$$
n_r = \frac{c}{v} = \frac{\varepsilon_0 \mu_0 \left[ 1 + (c \omega) \right]^2}{2}
$$

(4)

Equation (3) shows that $\omega/k_r = v$. Thus, $E/B = \omega/k_r = v$, i.e.,

$$
E = vB = \nu \mu H
$$

(5)

Then, Eq. (2) can be rewritten as follows

$$
W = \frac{1}{2} \varepsilon \nu^2 \mu^2 H^2 + \frac{1}{2} \mu H^2 = \frac{1}{2} \mu H^2 \left( \varepsilon \nu^2 \mu + \frac{1}{2} \mu H^2 \right)
$$

(6)

For $\sigma >> \omega \varepsilon$, Eq. (3) gives

$$
n_r^2 = \frac{c^2}{\nu^2} = \frac{\mu \sigma}{2 \omega c^2}
$$

(7)

Substitution of Eqs. (6) and (5) into Eq. (1) gives
It is known that the electrical conductivity is proportional to both the concentration and the mobility of the ions and the free electrons, and is expressed by

\[ \sigma = \rho_e \mu_e + \rho_i \mu_i \] (10)

where \( \rho_e \) and \( \rho_i \) express respectively the concentrations \( (\text{C/m}^3) \) of electrons and ions; \( \mu_e \) and \( \mu_i \) are respectively the mobilities of the electrons and the ions.

In order to calculate the electrical conductivity of the air inside the metallic tube, we first need to calculate the concentrations \( \rho_e \) and \( \rho_i \).

Since the number of atoms per \( m^3 \), \( n_a \), is given by

\[ n_a = \frac{N_0 \rho_i}{A_s} \] (11)

where \( N_0 = 6.02214129 \times 10^{26} \text{ atoms/kmole} \), is the Avogadro’s number; \( \rho_i \) is the matter density (in \( \text{kg/m}^3 \)) and \( A_s \) is the molar mass (\( \text{kg/kmole}^{-1} \)). Then, for \( \rho_i = \rho_{\text{air}} = 1 \times 10^4 \text{kg/m}^3 \) \( \left(6.62 \times 10^2 \text{ Torr}\right) \), Eq. (11) gives

\[ n_a = \frac{N_0 \rho_{\text{air}}}{A_{\text{air}}} \approx \frac{6.02214129 \times 10^{26} \times 1 \times 10^{-4}}{28.0134} = 2.15 \times 10^{21} \text{ atoms/m}^3 \] (12)

Using techniques of the Statistical Mechanics we can calculate the most probable number of ions, \( N_i \), in the volume \( V = \frac{4}{3} \pi \phi^2 h_\phi \) of air cylindrical tube, by means of the following expression [3].

\[ N_i = \frac{N}{S} a_i \] (13)

where \( N = n_a \times 1m^2 \) is the total number of atoms; \( S \) is the area total of the box \( (1m^2) \) and \( a_i \) is the area of the cell \( (h_\phi \phi) \). Therefore, Eq. (13) can be rewritten as follows

\[ N_i = n_a \frac{(1m^2)}{(1m^2)} (h_\phi \phi) \] (14)

For \( \phi = 1.6 \text{mm} \) and \( h_\phi = 12 \text{mm} \) Eq.(14) yields
Obviously, the number of free electrons will be equal the number of ions, thus we can write that
\[ N_e = N_i = 4.1 \times 10^{17} \text{ions} \] (15)
Now, we can calculate the concentrations \( \rho_e \) and \( \rho_i \) \( (C/m^3) \) of electrons and ions by means of the following expression
\[ \rho_e = \rho_i = \frac{eN_i}{V} = \frac{eN_i}{(\pi \phi^2) h_\phi} = 2.7 \times 10^7 C/m^3 \] (17)

This corresponds to a strong concentration level in the case of conducting materials. For these materials, at temperature of 300K, the mobilities \( \mu_e \) and \( \mu_i \) vary from \( 10^{-4} \) to \( 10^{-1} \) s V m\(^{-1}\). Assuming that \( \mu_e = \mu_i = 30 m^2 V^{-1}s^{-1} \) (Geometric mean of mobility level for conducting materials), the electrical conductivity of the air inside the tube is given by
\[ \sigma_{air} = \rho_e \mu_e + \rho_i \mu_i = 2(2.7 \times 10^7)(30) = \approx 2 \times 10^7 S.m^{-1} \] (18)

The pressure, \( \bar{P} \), exerted on the area \( S_\phi \), by of the air confined inside the tube, according to Eq.(1), is given by
\[ \bar{P} = \frac{m_g \bar{g}}{S_\phi} = \frac{2m_0 \bar{g}}{S_\phi h_\phi} = \frac{\chi \rho_{air} h_\phi \bar{g}}{S_\phi} \] (19)

Substitution of Eq.(9) into Eq. (19) gives
\[ \bar{P} = \left[ \rho_{air} - 2\sqrt{\left( \rho_{air} + \frac{\mu_{air} \sigma_{air}}{4\pi \mu_e^2} \right) H_{rms} - \rho_{air}} \right] h_\phi \bar{g} \] (20)

For \( f = 0.2Hz \), \( \sigma_{air} \approx 2 \times 10^7 S/m \), Eq.(20) gives
\[ \bar{P} = \left[ \rho_{air} - 2\sqrt{\left( \rho_{air} + 1.75 \times 10^{28} H_{rms} \right) - \rho_{air}} \right] h_\phi \bar{g} \] (21)

For \( \rho_{air} = 1 \times 10^{-3} kg.m^{-3} (6.62 \times 10^{-2} Torr) \), and \( H_{rms} = 7.96 \times 10^4 A/m \) \( [1 T] \), Eq. (21) gives
\[ \bar{P} = -0.0164 h_\phi \bar{g} \] (22)

Note the sign (-) in Eq. (13). It indicates that, in this case, the pressure \( \bar{P} \) acts on the contrary direction of the gravity \( \bar{g} \). This means that the pressure will be exerted on the upper surface of the cylindrical tube (See Fig.1). Thus, the cylindrical tube works as a gravitational micro-thruster, producing a thrust \( \bar{F} \), given by
\[ \bar{F} = \bar{P} S_\phi = -0.0164 h_\phi \bar{g} \] (23)

If \( h_\phi = 12 cm \), \( \phi = 1.6 mm \) and \( g = 9.81 m.s^{-2} \), then the intensity of the force \( \bar{F} \) is given by
\[ F = 3.89 \times 10^{-8} N \] (24)

In the case of a plate with \( N_\phi \) gravitational micro-thrusters, the Eq. (24) can be rewritten as follows
\[ F = 3.89 \times 10^{-8} N_\phi \] (25)
For example, if \( N_\phi = (560 \times 560) = 313600 \), then Eq. (25) gives
\[ F = 0.0122 N = 1.2 \times 10^{-3} kg f = 1.2 gf \] (26)

Assuming that, the 313600 metallic cylindrical tubes (external diameter= \( \phi_{ex} = 0.4 \text{mm} \) and height = \( h_\phi \)) are distributed into a square dielectric plate with sides \( l \), according to the pattern shown in Fig.2, then we can write that
\[ l = 2.2 \phi_{ex} \sqrt{N_\phi/S} \] (See Fig. 2). This means that the dielectric plate will have
\[ l = 1.10 m \] (27)
Figure 3 shows an experimental set-up in order to check the total thrust produced by the dielectric plate, with \( N \phi \) gravitational micro-thrusters, subjected to an oscillating magnetic field, given by:

\[
H_{rms} = (N_{\text{turns}}/y)_{rms} = (300/0.01)_{rms} = 3 \times 10^4 i_{rms}
\]

Coil: 300 turns (in X, in order to reduce the effect of distributed capacitance); # 14 AWG

\( N \phi \) Gravitational Micro-Thrusters (in white)
(Filled with air at 66.2 m Torr)
(Diameter \( \phi = 1.6 \text{mm} \), height \( h = 12 \text{cm} \),
Air density \( \rho \approx 1 \times 10^{-4} \text{kg/m}^3 \))

Fig. 3 – Schematic Diagram of the dielectric plate, with \( N \phi \) Gravitational Micro-Thrusters, on a balance, and inside a magnetic field produced by two external coils.

The mass of one dielectric plate (High-density polyethylene (HDPE), 970 kg/m\(^3\), 12.2 cm thickness (12.2 cm = \( h_\phi \) + 2 mm)) (See Fig 3), without the micro-thrusters is given by.

\[
m_{\text{mg}} = l^2(12.2 \text{cm})(970) = 143.19 \text{kg}
\]

The mass of one dielectric plate, 12.2 cm thickness, with \( N \phi \) cylindrical tubes, is given by

\[
m_{\text{mg}} = m_{\text{mg}} - N_{\phi}(970 \times \phi^2 \times h_\phi) = 28.51 \text{kg}
\]

The mass of one dielectric plate, 2 mm thickness (Bottom cover, see Fig.3), is

\[
m_{\text{mg}(2 \text{mm})} = l^2(2 \text{mm})(970) = 2.35 \text{kg}
\]

Thus, the total mass of the system is given by

\[
M_{\text{total}} = m_{\text{mg}} + m_{\text{mg}(2 \text{mm})} = 28.51 + 2.35 = 30.86 \text{kg}
\]

Therefore, the balance (see Fig.3) can have the following characteristics:

- **Maximum capacity**: 35 kg
- **Measuring accuracy**: 0.1 g

If the magnetic field \( H_{rms} \) is increased to \( 5.57 \times 10^4 \text{A/m} \) \((700T) \) \(* (\rho = 1 \times 10^{-4} \text{kg/m}^3)\), then Eq. (21) gives

\[
P = -9.66 \times 10^3 N/m^2
\]

Therefore, the thrust \( F_\phi \) produced by one micro-thruster (diameter \( \phi = 1.6 \text{mm} \) and height \( h = 12 \text{cm} \)), is given by

\[
F_\phi = PS_\phi = \pi \phi^2 P = 0.0194N
\]

Consequently, the total thrust produced by the system (one plate) with \( N_{\phi} = 313600 \) micro-thrusters will be given by

\[
F_{N_{\phi}} = N_{\phi} F_\phi = 6090.9N
\]

In practice, we can overlap several similar plates in order to increasing the total thrust. In this case, if the number of plates is \( N_{\text{plates}} \), the result is

\[
F_{\text{total}} = N_{\text{plates}} F_{N_{\phi}}
\]

For example, if number of overlapping plates (with \( N_{\phi} = 313600 \) micro-thrusters) is \( N_{\text{plates}} = 27 \), then the total thrust produced by the stack of plates (~3.4 cm total height) will be given by

\[
F_{\text{total}} \approx 164.4kN
\]

This thrust is of the order of the thrust of a fifth-generation jet fighter F-22 Raptor, which reaches 160,000N.

\* Recently, a magnetic field of 1200 T was generated by the electromagnetic flux-compression (EMFC) technique with a newly developed megagauss generator system [5].
It is important to note that the vertical direction \((-\bar{g})\) of the thrust produced by the plates can be turned of 90°, in order to produce horizontal displacements (See Fig.4 and Fig.5).

Fig. 4 - The vertical direction \((-\bar{g})\) of the thrust produced by a plate, with \(N_{\phi}\) gravitational micro-thrusters, can be turned of 90°, in order to produce horizontal displacements.

Fig. 5 – Horizontal displacement of \(F_x\) in several directions, simply rotating the plate with \(N_{\phi}\) Gravitational Micro-Thrusters.

Thus, gravitational micro-thrusters can be used to produce vertical or horizontal thrusts (in respect to \(\bar{g}\)). It is easy to see that, due to the magnitude of the thrust produced by these systems and their versatility, they can be can be used to move several types of vehicles.

References


APPENDIX A – Gravitational Thruster System (using plates with $N_{\phi}$ Gravitational Micro-Thrusters) to be used in aerospatial vehicles: the sub-systems S5 and S6 provide the vertical thrust (lift force). The sub-systems S1, S2, S3, S4 provide the horizontal thrusts in any direction. Note that the opposite thrusts (S1,S3), (S2,S4) can stop quickly the vehicle (see top view).

Top view

Side view
APPENDIX B – Ionization of the atoms of the Air inside the Micro-thrusters.

If $F_1 > F_o$ then $F_2 > F_o$ because $F_2 > F_1$. Under these circumstances, occurs the ionization of the atoms of the Air inside the micro-thrusters. This means that the condition for the ionization occurs is $F_1 > F_o$. By comparing the expressions of $F_1$ and $F_o$, we then conclude that

$$q > \left( \frac{r_1}{r_2 - r_1} \right)^2 e$$

(1)

However, if the number of micro-thrusters in each plate is $N_φ$, then the total charge, $Q$, to be provide by the capacitor is

$$Q = N_φ q > N_φ \left( \frac{r_1}{r_2 - r_1} \right)^2 e$$

(2)

If we use $N_c$ similar capacitors associated in parallel, then the total charge produced by them is given by

$$Q = N_c q = N_c CV$$

(3)

By comparing with Eq. (2) with Eq. (3), we conclude that

$$V > \left( \frac{N_φ}{N_c} \right) \frac{r_1}{r_2 - r_1} \frac{e}{C}$$

(4)

For $N_φ = 313600$; $N_c = 10$; $r_2 - r_1 \equiv 10^{-10} m \Rightarrow r_2 \equiv r_1 \equiv \phi/2$ and $C = 8000 \mu F - 75V$, the Eq. (4) gives

$$V > 41.1 \text{ volts}$$

(5)